

MATRIX METHODS IN STRUCTURAL DYNAMICS

ABSTRACT

PARTIAL FULFILLMENT IN STRUCTURAL DYNAMICS
by

Charan Phimphilai

Master of Science

Youngstown State University

Submitted in Partial Fulfillment of the Requirements

The purpose for the Degree of Master of Science in FORTAN IV
usable set of computer programs which are associated with problems
in the Civil Engineering
structural dynamics. The program is accompanied by a
review of the matrix theory, a complete flow chart, a
print out of the actual computer program, and a sample

Paul Bellini *23 Jan 78*

Advisor _____ Date

Len Randal *Feb. 6 1978*

Dean of the Graduate School Date

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problem, the characteristic value problem, the
characteristic vector problem, the normalization
problems and the Cholesky Triangularization method.
In the second section a computer program using finite

YOUNGSTOWN STATE UNIVERSITY

* number in parentheses January, 1978 literature cited
in the bibliography

ABSTRACT

MATRIX METHODS IN STRUCTURAL DYNAMICS.

subject to Charan Phimphilai

Master of Science

Youngstown State University

The purpose of this thesis is to formulate a usable set of computer programs written in FORTRAN IV computer language which are associated with problems that arise in the field of matrix operations in structural dynamics. Each program is accompanied by a review of the matrix theory, a complete flow chart, a print out of the actual computer program, and a sample example to illustrate the results.

This work is divided into two distinct parts. The first section includes a series of programs which analyse the determinant problem, the matrix inversion problem, the characteristic value problem, the characteristic vector problem, the normalization problem and the Cholesky Triangularization method.^{(1)*} In the second section a computer program using finite

* number in parenthesis refers to literature cited in the bibliography

ACKNOWLEDGEMENTS

difference techniques is written to determine the dynamic response of a lumped-mass structural system subject to external time-varying loading conditions.

The advantage of the latter method of analysis is that it completely eliminates the classical approach to the solution of the problem which includes both the necessity of computing the natural frequencies and modal shapes of the free vibration problem, and the utilization of a series-type integral solution for the problem.

ACKNOWLEDGEMENTS

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Great appreciation is given to my parents Mr. & Mrs. C. Phimphilai for supporting my studies.

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LIST OF SYMBOLS

SYMBOLS	DEFINITION
[A]	Symmetric matrix
[F]	Force matrix
[I]	Identity matrix
[K]	Elastic bending stiffness matrix
[L]	Lower triangular matrix
[M]	Mass matrix
[U]	Upper triangular matrix
{s}	Associated displacement vector
{u}	Eigen Vector
{x}	Displacement Vector
{\dot{x}}	Velocity Vector
{\ddot{x}}	Acceleration Vector
{x}^{(0)}	Initial displacement vector
{\dot{x}}^{(0)}	Initial velocity vector
{\ddot{x}}^{(0)}	Initial acceleration vector
{y}	Associated displacement vector
A	Cross - sectional area of member
E	Young's modulus of elasticity
L	Length of member
P	Axial force
V	Shear force
W	Weight of the tributary wall areas

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In addition, one of the first basic text books relating to the matrix methods in civil engineering by Pipes⁽¹⁾. This text book covers the basic forms of matrix operation with applications to elasticity, dynamics, vibrations, and structural analysis. A second book by Pipes⁽²⁾ published in 1969 presents a series of actual computer programs which may be utilized to efficiently solve many of the usual problems associated with

* number in parenthesis refers to literature cited in the bibliography

CHAPTER I

INTRODUCTION

The introduction of the high speed electronic computer to field of numerical computation has revolutionized the approach to the analytical solution of many complicated problems. It has become particularly valuable to the field of engineering where specially prepared computer programs have been developed to aid in the solution of problems in structural analysis, stress analysis, surveying, fluid mechanics, machine design, vibrations and structural dynamics.

In most of the latter computer programs, the fundamental mathematical operations present are those related to matrix operations, including both matrix algebra and matrix calculus. From an engineering standpoint, one of the first basic text books related to the matrix methods is that written by Pipes^{(1)*}. This text book covers the basic forms of matrix operation with applications to elasticity, dynamics, vibrations, and structural analysis. A second book by Pipes⁽²⁾ published in 1969 presents a series of actual computer programs which may be utilized to efficiently solve many of the usual problems associated with

* number in parenthesis refers to literature cited in the bibliography

important matrix operations.

The purpose of this thesis is to develop a series of programs for suitable use on IBM 360-70 which is available at Youngstown State University which contain the matrix operation applicable to the solution of a typical engineering problem.

For each program formulated, a flow chart, a complete computer program in FORTRAN IV, and a sample of example illustrating the problem is presented for clarity and ease of interpretation.

The following list of computer programs are formulated:

- 1) Determinant Evaluation of a Matrix
- 2) Inversion Evaluation of a Matrix
- 3) Characteristic Equation Evaluation of a Matrix
- 4) Characteristic Value and Characteristic Vector Evaluation of a Matrix
- 5) Cholesky Transformation Evaluation for a Matrix.

The first four programs are standard problems in matrix operations which are essential to all matrix analysis procedures. The last program is a more recently developed technique in which a matrix is replaced by the product of an upper triangular and

a lower triangular matrix. This technique is summarized by Westlake⁽³⁾ and is specialized for the case of a symmetric matrix. Application of this technique to real engineering problems has been summarized by Parsons⁽⁴⁾ in a master thesis at Youngstown State University.

The second section of this thesis presents a computer program solution for the analysis of a typical problem in Structural Dynamics. This solution consists of determining the dynamic response of a single-bay, multi-story, planar frame subjected to time varying forces. The method combines the use of finite difference techniques as reviewed by Rogers⁽⁵⁾ simultaneously with matrix operations. The problem includes the modeling of the structural frame into a lumped-mass and spring mechanical system which generates a set of linear, coupled, total differential equation. The solution of these equations using classical scalar manual techniques is summarized by Fertis⁽⁶⁾.

The computer solution developed in this work offers an efficient and economical means of determining the response of the structure for a variety of dynamic

loading conditions and at the same time minimizes the time required to obtain these solutions.

3.2. DETERMINANT EVALUATION

The determinant evaluation process is based on the method of Chio^{(2)*}. The idea of this method is to reduce the order of the determinant from higher order to lower order until a (1×1) determinant is obtained which gives the actual value of the original determinant. This method is illustrated using the following (4×4) determinant.

$$D^* = \begin{vmatrix} a_{11} & a_{12} & a_{13} & a_{14} \\ a_{21} & a_{22} & a_{23} & a_{24} \\ a_{31} & a_{32} & a_{33} & a_{34} \\ a_{41} & a_{42} & a_{43} & a_{44} \end{vmatrix} \quad (2.1.1)$$

To reduce the order of the determinant from a (4×4) up a (3×3) , any element of the determinant, say element (1,1), is made equal to unity by dividing the

* Number in parenthesis refers to literature cited in the bibliography.

CHAPTER II

GENERAL PROGRAMS

2.1 FORTRAN Program for Determinant Evaluation

This Determinant Evaluation program is based on the method of Chio^{(2)*}. The idea of this method is to reduce the order of the determinant from higher order to lower order until a (1 x 1) determinant is obtained which gives the actual value of the original determinant. This method is illustrated using the following (4 x 4) determinant.

$$D = \begin{vmatrix} a_{11} & a_{12} & a_{13} & a_{14} \\ a_{21} & a_{22} & a_{23} & a_{24} \\ a_{31} & a_{32} & a_{33} & a_{34} \\ a_{41} & a_{42} & a_{43} & a_{44} \end{vmatrix}$$

To reduce the order of the determinant from a (4 x 4) to a (3 x 3), any element of the determinant, say element (1,1), is made equal to unity by dividing the

* number in parenthesis refers to literature cited in the bibliography

first row through by a_{11} , yielding

$$D = a_{11} \begin{vmatrix} 1 & \frac{a_{12}}{a_{11}} & \frac{a_{13}}{a_{11}} & \frac{a_{14}}{a_{11}} \\ a_{21} & a_{22} & a_{23} & a_{24} \\ a_{31} & a_{32} & a_{33} & a_{34} \\ a_{41} & a_{42} & a_{43} & a_{44} \end{vmatrix} \quad (2.1.2)$$

Setting $a'_{12} = a_{12}/a_{11}$
 $a'_{13} = a_{13}/a_{11}$
 $a'_{14} = a_{14}/a_{11}$

multiplying the first row by a_{21} and subtracting from the second row, then multiplying the first row by a_{31} and subtracting from the third row and, finally multiplying the first row by a_{41} and subtracting from the fourth row, gives

$$D' = a_{11} \begin{vmatrix} 1 & a'_{12} & a'_{13} & a'_{14} \\ 0 & a_{22} - a_{21}a'_{12} & a_{23} - a_{21}a'_{13} & a_{24} - a_{21}a'_{14} \\ 0 & a_{32} - a_{31}a'_{12} & a_{33} - a_{31}a'_{13} & a_{34} - a_{31}a'_{14} \\ 0 & a_{42} - a_{41}a'_{12} & a_{43} - a_{41}a'_{13} & a_{44} - a_{41}a'_{14} \end{vmatrix}$$

(2.1.3)

The value of this determinant now becomes

$$D'' = a_{11} (-1)^{1+1} \begin{vmatrix} a''_{11} & a''_{12} & a''_{13} \\ a''_{21} & a''_{22} & a''_{23} \\ a''_{31} & a''_{32} & a''_{33} \end{vmatrix} \quad (2.1.4)$$

where $(-1)^{1+1}$ is the sign of element $(1,1)$, and

$$a''_{11} = a_{22} - a_{21}a'_{12} \quad a''_{12} = a_{23} - a_{21}a'_{13} \quad a''_{13} = a_{24} - a_{21}a'_{14}$$

$$a''_{21} = a_{32} - a_{31}a'_{12} \quad a''_{22} = a_{33} - a_{31}a'_{13} \quad a''_{23} = a_{34} - a_{31}a'_{14}$$

$$a''_{31} = a_{42} - a_{41}a'_{12} \quad a''_{32} = a_{43} - a_{41}a'_{13} \quad a''_{33} = a_{44} - a_{41}a'_{14}$$

Thus, Equation (2.1.4) is a determinant of order (3×3) . By repeating this operation, the determinant is reduced to a (2×2) and finally a (1×1) with the multipliers $a_{11} a''_{11} a'''_{11} \dots$, from which the value of the original (4×4) determinant is obtained.

The basic flow chart of this program is shown in Figure 1.

Figure 1 Flow chart of Determinant program

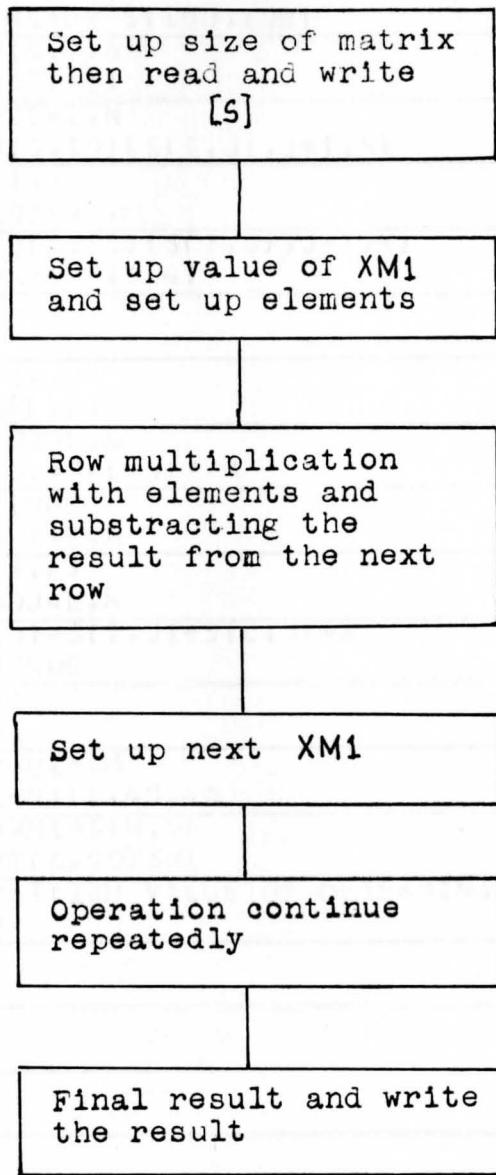


Figure 1 Flow chart of Determinant program

```
C      NAME OF THE PROGRAM=DETER
C      TO DETERMINE THE DETERMINANT OF THE MATRIX N*N
C      DIMENSION S(100,100)
      READ(5,2)N
      FORMAT(1I1)
      DO 90I=1,N
      READ(5,10)(S(I,J),J=1,N)
      90  CONTINUE
      DO 100I=1,N
      100 WRITE(6,10)(S(I,J),J=1,N)
      10  FORMAT(8F10.4)
      K=2
      L=1
      XM1=1.
      11  XM=S(L,L)
      DO 20J=L,N
      S(L,J)=S(L,J)/XM
      20  CONTINUE
      DO 30I=K,N
      X=S(I,L)
      DO 30J=L,N
      S(I,J)=S(I,J)-S(L,J)*X
      30  CONTINUE
      L=L+1
      K=K+1
      XM1=XM1*X
      IF(L-N)11,40,40
      40  XM1=XM1*S(N,N)
      WRITE(6,50)XM1
      50  FORMAT(22H VALUE OF DETERMINANT= E10.3)
      STOP
      END
```

VS LOADER
10

OPTIONS USED - PRINT,MAP,LET,CALL,NORES,NOTERM,SIZE=102400,NAME=**GO

NAME	TYPE	ADDR	NAME	TYPE	ADDR	NAME	TYPE	ADDR
MAIN	SD	150810	IHN ECOMH*	SD	15A883	IBCOM# *	LR	15A8E4
IHN COMH2*	SD	15B848	SEQDASD *	LR	15BC76	IHN FCVTH*	SD	15BF60
FCVLOUTP*	LR	15C09A	FCVZOUTP*	LR	15C1F6	FCVIQUTP*	LR	15C59E
INT6SWCH*	LR	15CA5C	IHN EFIOS*	SD	15CADO	FIOCS# *	LR	15CADO
IHN EFNTH*	SD	15E080	ARITH# *	LR	15EC80	ADJSWTCH*	LR	15E41C
ERRMON *	LR	15E930	IHN ERRE *	LR	15E948	IHN FCNCO*	SD	15EF30
FQCONI# *	LR	15F3D8	IHN UATBL*	SD	15F6C0	IHN ETRCH*	SD	15F948
IHN FTEN *	SD	15FBF0	FTEN# *	LR	15FBF0			

TOTAL LENGTH F578

ENTRY ADDRESS 150810

15.0000	1.0000	2.0000	-3.0000
5.0000	6.0000	4.0000	4.0000
-10.0000	-3.0000	2.0000	1.0000
-5.0000	3.0000	4.0000	0.0

VALUE OF DETERMINANT=-0.182E+04

OPTIONS USED = PRINT,MAP,LET,CALL,NORES,NOTERM,SIZE=102400,NAME=***GO

NAME	TYPE	ADDR	NAME	TYPE	ADDR	NAME	TYPE	ADDR
MAIN	SD	180810	IHNCOMH*	SD	18A888	IBCOM# *	LR	18A8E4
IHNCOMH2*	SD	188848	SEQDASD *	LR	18B076	IHNFCVTH*	SD	18BE60
FCVLOUTP*	LR	18C09A	FCVZOUTP*	LR	18C1F6	FCVIQUTP*	LR	18C59E
INT6SWCH*	LR	18CA50	IHNFIOS*	SD	18CADO	FIDCS# *	LR	18CAD0
IHNFFENTH*	SD	18E080	ARITH# *	LR	18E080	ADJISWTCH*	LR	18F41C
ERRMON *	LR	18E930	IHNERR*	LR	18E948	IHNFCONO*	SD	18EF30
FOCONI# *	LR	18F3D8	IHNNUATBL*	SD	18F6C0	IHNTRCH*	SD	18F948
IHNFTEN *	SD	18FBF0	FTEN# *	LR	18FBF0			

TOTAL LENGTH = F578

ENTRY ADDRESS 180810

1.0000 2.0000 3.0000

3.0000 4.0000 6.0000

2.0000 1.0000 1.0000

VALUE OF DETERMINANT = 0.100E+01

2.2 Program for the Inverse Matrix Evaluation

This matrix inversion program is formulated using the augmented matrix technique which is based on the Gauss-Jordan⁽²⁾ method of solving simultaneous equations. This method involves the use of a unit matrix of the same order as the original matrix attached to the right hand side of the original matrix producing an ($n \times 2n$) matrix. This new matrix is the augmented matrix in the form of Equation (2.2.1). Then, performing the proper matrix row operations, the original matrix is reduced to a unit matrix. The same row operations when applied to the attached unit matrix transform it into the inverse matrix

$$\left[\begin{array}{cccc|ccccc} a_{11} & a_{12} & a_{13} & \dots & a_{1n} & 1 & 0 & 0 & \dots & 0 \\ a_{21} & a_{22} & a_{23} & \dots & a_{2n} & 0 & 1 & 0 & \dots & 0 \\ a_{31} & a_{32} & a_{33} & \dots & a_{3n} & 0 & 0 & 1 & \dots & 0 \\ \vdots & \vdots & \vdots & & \vdots & \vdots & \vdots & \vdots & & \vdots \\ \vdots & \vdots & \vdots & & \vdots & \vdots & \vdots & \vdots & & \vdots \\ \vdots & \vdots & \vdots & & \vdots & \vdots & \vdots & \vdots & & \vdots \\ a_{n1} & a_{n2} & a_{n3} & \dots & a_{nn} & 0 & 0 & 0 & \dots & 1 \end{array} \right]$$

(2.2.1)

The general procedure includes dividing the first row by the leading coefficient, multiplying the first row by the leading coefficient of the second row, and then subtracting it from the second row. This procedure repeated for the third row through the n^{th} row yields

$$\left[\begin{array}{cccc|ccc} 1 & 0 & 0 & \dots & 0 & b_{11} & b_{12} & b_{13} & \dots & b_{1n} \\ 0 & 1 & 0 & \dots & 0 & b_{21} & b_{22} & b_{23} & \dots & b_{2n} \\ 0 & 0 & 1 & \dots & 0 & b_{31} & b_{32} & b_{33} & \dots & b_{3n} \\ \vdots & \vdots & \vdots & & \vdots & & & & & \\ \vdots & \vdots & \vdots & & \vdots & & & & & \\ 0 & 0 & 0 & \dots & 1 & b_{n1} & b_{n2} & b_{n3} & \dots & b_{nn} \end{array} \right] \quad (2.2.2)$$

The square matrix $[B]$ in the right hand side of the Equation (2.2.2) is the inverse of the matrix $[A]$. The basic flow chart for the inversion matrix program is shown in Figure 2.

Figure 2 Flow chart of Inversion program

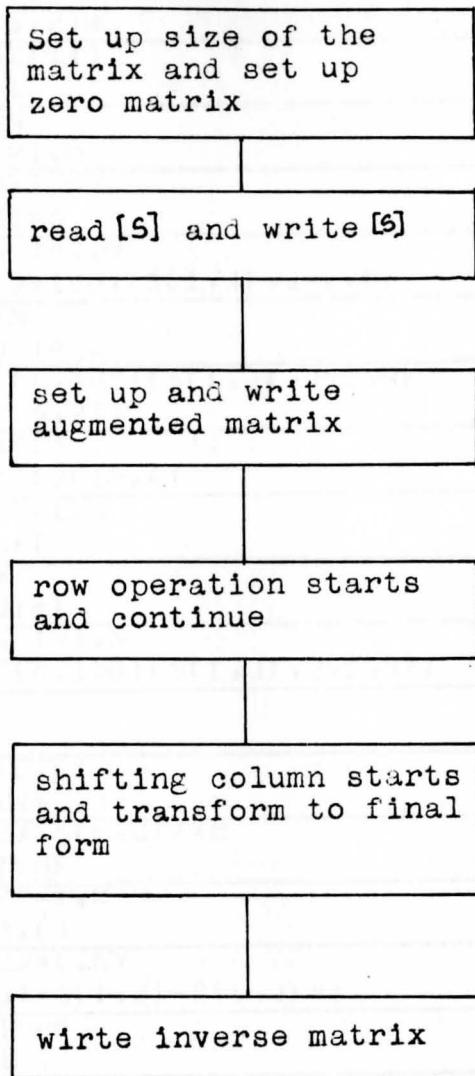


Figure 2 Flow chart of Inversion program

C MATRIX INVERSION BY AUGMENTED MATRIX METHOD

```

DIMENSION S(50,50)
READ(5,2)N
2 FORMAT(1I1)
NX=N+1
NY=2*N
DO 5 I=1,N
DO 5 J=1,NY
5 S(I,J)=0
DO 10 I=1,N
READ(5,100)(S(I,J),J=1,N)
10 CONTINUE
DO 20 I=1,N
20 WRITE(6,100)(S(I,J),J=1,N)
WRITE(6,21)
21 FORMAT('      ')
100 FORMAT(2F10.4)
DO 30 I=1,N
NXX=N+I
J=NXX
30 S(I,J)=1
DO 11 I=1,N
11 WRITE(6,100)(S(I,J),J=1,NY)
L=1
K=2
31 XM=S(L,L)
DO 40 J=L,NY
S(L,J)=S(L,J)/XM
40 CONTINUE
DO 50 I=K,N
X=S(I,L)
DO 50 J=L,NY
S(I,J)=S(I,J)-S(L,J)*X
50 CONTINUE
L=L+1
K=K+1
IF(L-N)31,31,51
51 L=N
52 LZ=L-1
DO 60 K=1,LZ
I=L-K
Y=S(I,L)
DO 60 J=L,NY
S(I,J)=S(I,J)-S(L,J)*Y
60 CONTINUE
L=L-1
IF(L-1)61,61,52
61 WRITE(6,200)((S(I,J),J=NX,NY),I=1,N)
WRITE(6,21)
200 FORMAT(15H INVERSE MATRIX/(3X1P4E20.3))
STOP
END

```

VS LOADER

16

OPTIONS USED - PRINT,MAP,LET,CALL,NORES,NOTERM,SIZE=102400,NAME==**GO

NAME	TYPE	ADDR	NAME	TYPE	ADDR	NAME	TYPE	ADDR
MAIN	SD	150810	IHNECOMH*	SD	15AB90	IBCOM#	*	LR 15ABBC
IHNCOMH2*	SD	15BB20	SEQDASD *	LR	15BF4E	IHNFCVTB*	SD	15C238
FCVLOUTP*	LR	15C372	FCVZOUTP*	LR	15C4CE	FCVIOUTP*	LR	15C876
INT6SWCH*	LR	15CD28	IHNEFIOS*	SD	15CDA8	FIOCS#	*	LR 15CDA8
IHNEFNTH*	SD	15E358	ARITH#	*	LR 15E358	ADJSWTCH*	LR	15E6F4
ERRMON *	LR	15EC08	IHNERR*	LR	15EC20	IHNFCONO*	SD	15F208
FQCONI# *	LR	15F6B0	IHNUTABL*	SD	15F998	IHNTRCH*	SD	15FC20
IHNFTEN *	SD	15FEC8	FTEN#	*	LR 15FEC8			

TOTAL LENGTH F850
ENTRY ADDRESS 150810

2.0000	1.0000	1.0000
1.0000	3.0000	1.0000
1.0000	1.0000	4.0000

2.0000	1.0000	1.0000	1.0000	0.0	0.0
1.0000	3.0000	1.0000	0.0	1.0000	0.0
1.0000	1.0000	4.0000	0.0	0.0	1.0000

INVERSE MATRIX

6.471E-01	-1.765E-01	-1.176E-01
-1.765E-01	4.118E-01	-5.882E-02
-1.176E-01	-5.882E-02	2.941E-01

OPTIONS USED - PRINT,MAP,LET,CALL,NORES,NOTERM,SIZE=102400,NAME=**GO

NAME	TYPE	ADDR	NAME	TYPE	ADDR	NAME	TYPE	ADDR
MAIN	SD	150810	IHN ECOMH*	SD	15AB90	IB COM# *	LR	15AB
IHN COMH2*	SD	15BB20	SEQDASD *	LR	15BF4E	IHN FCVTH*	SD	15C2
FCVL OUTP*	LR	15C372	FCVZ OUTP*	LR	15C4CE	FCV IOUTP*	LR	15C8
INT6SWCH*	LR	15CD28	IHN EFIO S*	SD	15CDA8	FIOCS# *	LR	15CD
IHN EFNT H*	SD	15E358	ARI TH# *	LR	15E358	ADJSWTC H*	LR	15E6
ERRMON *	LR	15EC08	IHN ERRE *	LR	15EC20	IHN FCONC*	SD	15F2
FQCONI# *	LR	15F6B0	IHN UATBL*	SD	15F998	IHN ET RCH*	SD	15FC
IHN FTEN *	SD	15FEC8	FTEN# *	LR	15FEC8			

TOTAL LENGTH F850
ENTRY ADDRESS 150810

15.0000	1.0000	2.0000	-3.0000				
5.0000	6.0000	4.0000	4.0000				
-10.0000	-3.0000	2.0000	1.0000				
-5.0000	3.0000	4.0000	0.0				

15.0000	1.0000	2.0000	-3.0000	1.0000	0.0	0.0
5.0000	6.0000	4.0000	4.0000	0.0	1.0000	0.0
-10.0000	-3.0000	2.0000	1.0000	0.0	0.0	1.0000
-5.0000	3.0000	4.0000	0.0	0.0	0.0	0.0

INVERSE MATRIX

4.615E-02	3.077E-02	1.538E-02
-8.791E-02	-1.099E-02	-2.198E-01
1.236E-01	4.670E-02	1.841E-01
-4.945E-02	1.813E-01	1.264E-01

2.3 FORTRAN Program for the Characteristic
Equation Problem

The method of Danilevsky⁽²⁾ for obtaining the characteristic equation is based on the reduction of matrix ,

$$[A] = \begin{bmatrix} a_{11} & a_{12} & a_{13} & \dots & a_{1n} \\ a_{21} & a_{22} & a_{23} & \dots & a_{2n} \\ \cdot & \cdot & \cdot & & \\ \cdot & \cdot & \cdot & & \\ \cdot & \cdot & \cdot & & \\ a_{n1} & a_{n2} & a_{n3} & \dots & a_{nn} \end{bmatrix} \quad (2.3.1)$$

whose characteristic equation is desired, to matrix [P] of the form

$$[P] = \begin{bmatrix} p_1 & p_2 & p_3 & \dots & p_{n-1} & p_n \\ 1 & 0 & 0 & \dots & 0 & 0 \\ 0 & 1 & 0 & \dots & 0 & 0 \\ \cdot & \cdot & \cdot & & & \\ \cdot & \cdot & \cdot & & & \\ \cdot & \cdot & \cdot & & & \\ 0 & 0 & 0 & \dots & 1 & 0 \end{bmatrix} \quad (2.3.2)$$

The elements $p_1, p_2, p_3 \dots, p_n$ represent the coefficients of the characteristic equation of matrix $[P]$. The process is accomplished by a series of similarity transformations. Since the characteristic equations of similar matrices are identical, the characteristic equations of matrices $[A]$ and $[P]$ are the same. Coefficients of the characteristic polynomial of matrix $[P]$ is given by the determinant

$$D(\lambda) = \left| [P] - \lambda[I] \right| = \begin{vmatrix} p_1 - \lambda & p_2 & p_3 & \dots & p_n \\ 1 & -\lambda & 0 & \dots & 0 \\ 0 & 1 & -\lambda & \dots & 0 \\ \vdots & & & & \vdots \\ 0 & 0 & 0 & & -\lambda \end{vmatrix} \quad (2.3.3)$$

which expanded in terms of the elements of the first row, gives

$$D(\lambda) = (-1)^n (\lambda^n - p_1 \lambda^{n-1} - p_2 \lambda^{n-2} - \dots - p_n) = 0$$

where $D(\lambda)$ is the desired characteristic polynomial of both matrix $[A]$ and $[P]$ and n is the order of the square matrix $[A]$.

The transformation from matrix [A] to matrix [P]
is performed on the following (4 x 4) matrix

$$[A] = \begin{bmatrix} a_{11} & a_{12} & a_{13} & a_{14} \\ a_{21} & a_{22} & a_{23} & a_{24} \\ a_{31} & a_{32} & a_{33} & a_{34} \\ a_{41} & a_{42} & a_{43} & a_{44} \end{bmatrix} \quad (2.3.4)$$

Matrix [M] is first formed from the elements of the
fourth row of matrix [A] as

$$[M] = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ -\frac{a_{41}}{a_{43}} & -\frac{a_{42}}{a_{43}} & -\frac{1}{a_{43}} & \frac{a_{44}}{a_{43}} \\ 0 & 0 & 1 & 0 \end{bmatrix} \quad (2.3.5)$$

The product of $[A]$ and $[M]$ become

$$[A][M] = \begin{bmatrix} a_{11} & a_{12} & a_{13} & a_{14} \\ a_{21} & a_{22} & a_{23} & a_{24} \\ a_{31} & a_{32} & a_{33} & a_{34} \\ a_{41} & a_{42} & a_{43} & a_{44} \end{bmatrix} \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ -\frac{a_{41}-a_{42}}{a_{43}} & \frac{1}{a_{43}} & -\frac{a_{44}}{a_{43}} \\ 0 & 0 & 0 & 1 \end{bmatrix}$$

or

$$[A][M] = \begin{bmatrix} b_{11} & b_{12} & b_{13} & b_{14} \\ b_{21} & b_{22} & b_{23} & b_{24} \\ b_{31} & b_{32} & b_{33} & b_{34} \\ b_{41} & b_{42} & b_{43} & b_{44} \end{bmatrix} \quad (2.3.6)$$

where

$$b_{11} = a_{11} - \frac{a_{13}a_{14}}{a_{43}} \quad b_{12} = a_{12} - \frac{a_{13}a_{12}}{a_{43}} \quad b_{41} = 0$$

$$b_{13} = a_{23}/a_{43} \quad b_{14} = -a_{13}a_{44}/a_{43} \quad b_{42} = 0$$

$$b_{21} = a_{21} - \frac{a_{23}a_{41}}{a_{43}} \quad b_{22} = a_{22} - \frac{a_{23}a_{42}}{a_{43}} \quad b_{43} = 1$$

$$b_{23} = a_{23}/a_{43} \quad b_{24} = a_{24} - \frac{a_{23}a_{44}}{a_{43}} \quad b_{44} = 0$$

$$b_{31} = a_{31} - \frac{a_{33}a_{41}}{a_{43}} \quad b_{32} = a_{32} - \frac{a_{33}a_{42}}{a_{43}}$$

$$b_{33} = a_{33}/a_{43} \quad b_{34} = a_{34} - \frac{a_{33}a_{44}}{a_{43}}$$

The matrix of Equation (2.3.6) is not yet similar to matrix $[A]$. It is made similar to $[A]$ by premultiplying Equation (2.3.6) by the inverse of matrix $[M]$ which is obtained from Equation (2.3.5) as

$$[M]^{-1} = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ a_{41} & a_{42} & a_{43} & a_{44} \\ 0 & 0 & 0 & 0 \end{bmatrix} \quad (2.3.7)$$

Thus, the similarity transformation of $[A]$ takes the form

$$[C] = [M]^{-1} [A] [M] \quad (2.3.8 \text{ a})$$

$$[C] = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ a_{41} & a_{42} & a_{43} & a_{44} \\ 0 & 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} a_{11} & a_{12} & a_{13} & a_{14} \\ a_{21} & a_{22} & a_{23} & a_{24} \\ a_{31} & a_{32} & a_{33} & a_{34} \\ a_{41} & a_{42} & a_{43} & a_{44} \end{bmatrix} \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ -\frac{a_{41}}{a_{43}} & -\frac{a_{42}}{a_{43}} & \frac{1}{a_{43}} & -\frac{a_{44}}{a_{43}} \\ 0 & 0 & 0 & 1 \end{bmatrix}$$

$$[C] = \begin{bmatrix} C_{11} & C_{12} & C_{13} & C_{14} \\ C_{21} & C_{22} & C_{23} & C_{24} \\ C_{31} & C_{32} & C_{33} & C_{34} \\ 0 & 0 & 1 & 0 \end{bmatrix} \quad (2.3.8 \text{ b})$$

The process is continued in reducing the matrix of Equation (2.4.8b) to the form

$$[D] = \begin{bmatrix} 1 & 0 & 0 & 0 \\ C_{31} & C_{32} & C_{33} & C_{34} \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} C_{11} & C_{12} & C_{13} & C_{14} \\ C_{21} & C_{22} & C_{23} & C_{24} \\ C_{31} & C_{32} & C_{33} & C_{34} \\ 0 & 0 & 1 & 0 \end{bmatrix} \begin{bmatrix} 1 & 0 & 0 & 0 \\ -\frac{C_{31}}{C_{32}} & \frac{1}{C_{32}} & -\frac{C_{33}}{C_{32}} & -\frac{C_{34}}{C_{32}} \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix}$$

$$[D] = \begin{bmatrix} d_{11} & d_{12} & d_{13} & d_{14} \\ d_{21} & d_{22} & d_{23} & d_{24} \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \end{bmatrix} \quad (2.3.9)$$

The process is repeated a step further which reduces the (4×4) matrix of Equation (2.3.4) to the form of Equation (2.3.2) which simplifies to

$$[P] = \begin{bmatrix} p_1 & p_2 & p_3 & p_4 \\ 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \end{bmatrix} \quad (2.3.10)$$

where the p's represent the coefficients of the characteristic polynomial of Equation (2.3.4). Then, the characteristic polynomial equation is written as

$$D(\lambda) = (-1)^n (\lambda^n - p_1 \lambda^{n-1} - p_2 \lambda^{n-2} - p_3 \lambda^{n-3} - p_4) = 0$$

The flow chart of the computer program for the characteristic equation is shown in Figure 3.



Figure 3 Flow chart of Characteristic Equation

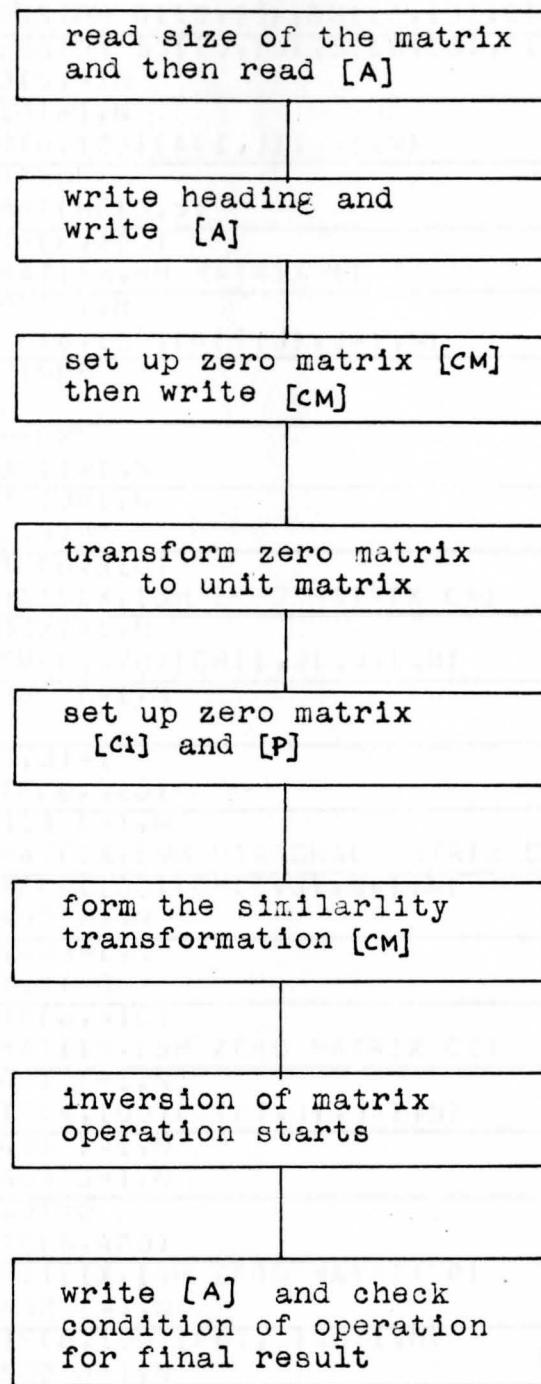


Figure 3 Flow chart of Characteristic Equation

```

C      THIS PROGRAM USE TO COMPUTE THE CHARACTERISTIC EQUATION
C      BY USING A. M. DANILEVSKY METHOD
DIMENSION A(10,10),CM(10,10),CI(10,10),P(10,10)
DIMENSION SS(10,10),S(10,10),R(10,10)
READ(5,*)N
DO 10 I=1,N
    READ(6,100)(A(I,J),J=1,N)
10  CONTINUE
100  FORMAT(8F10.5)
    WRITE(6,110)
110  FORMAT(1X,9H MATRIX A)
    DO 20 I=1,N
        WRITE(6,100)(A(I,J),J=1,N)
20  CONTINUE
LX=1
LX=N-LX
997  DO 301 I=1,N
    DO 301 J=1,N
301  CM(I,J)=0
    WRITE(6,310)
310  FORMAT(1X,15H ZERO MATRIX CM)
    DO 302 I=1,N
302  WRITE(6,100)(CM(I,J),J=1,N)
    DO 300 I=1,N
        J=I
300  CM(I,J)=1
        WRITE(6,320)
320  FORMAT(1X,19H DIAGONAL MATRIX CM,/)
303  WRITE(6,100)(CM(I,J),J=1,N)
    DO 400 I=1,N
        DO 400 J=1,N
400  CI(I,J)=0
        WRITE(6,410)
410  FORMAT(1X,15H ZERO MATRIX CI)
    DO 401 I=1,N
401  WRITE(6,100)(CI(I,J),J=1,N)
    DO 403 I=1,N
        DO 403 J=1,N
403  P(I,J)=0
        WRITE(6,420)
420  FORMAT(1X,14H ZERO MATRIX P)
    DO 402 I=1,N
402  WRITE(6,100)(P(I,J),J=1,N)
    DO 500 J=1,N
        IF(J-LX)520,530,520
520  CM(LX,J)=-A(LX+1,J)/A(LX+1,LX)
    GO TO 500
530  CM(LX,J)=1/A(LX+1,LX)
500  CONTINUE
    WRITE(6,510)
510  FORMAT(1X,14H NEW MATRIX CM)
    DO 600 I=1,N
600  WRITE(6,100)(CM(I,J),J=1,N)
    DO 700 I=1,N

```

```

DO 710 J=1,N
S(I,J)=CM(I,J)
710 CONTINUE
700 CONTINUE
WRITE(6,540)
540 FORMAT(1X,13H NEW MATRIX S)
DO 720 I=1,N
720 WRITE(6,100)(S(I,J),J=1,N)
NX=N+1
NY=2*N
DO 5 I=1,N
DO 5 J =1,N
5 S(I,J)=0
DO 32I=1,N
DO 32J=NX,NY
32 S(I,J)=0
DO 6 I=1,N
DO 7 J=1,N
7 S(I,J)=CM(I,J)
6 CONTINUE
CONTINUE
WRITE(6,540)
DO 30I=1,N
NXX=N+I
J=NXX
30 S(I,J)=1
DO 11 I=1,N
11 WRITE(6,100)(S(I,J),J=1,NY)
L=1
K=2
31 XM=S(L,L)
DO 40J=L,NY
S(L,J)=S(L,J)/XM
40 CONTINUE
DO 50I=K,N
X=S(I,L)
DO 50J=L,NY
S(I,J)=S(I,J)-S(L,J)*X
50 CONTINUE
L=L+1
K=K+1
IF(L-N)31,31,51
51 L=N
52 LZ=L-1
DO 60K=1,LZ
I=L-K
Y=S(I,L)
DO 60J=L,NY
S(I,J)=S(I,J)-S(L,J)*Y
60 CONTINUE
L=L-1
IF(L-1)61,61,52
61 WRITE(6,200)((S(I,J),J=NX,NY),I=1,N)
200 FORMAT(15H INVERSE MATRIX/(3X1P4E20.6))
DO 555 I=1,N

```

```
DO 555 J=1,N  
555 P(I,J)=0  
DO 666 M=1,N  
DO 70I=1,N  
DO 80K=1,N  
80 P(M,I)=A(M,K)*CM(K,I)+P(M,I)  
70 CONTINUE  
666 CONTINUE  
WRITE(6,222)  
222 FORMAT(15H PRODUCT MATRIX)  
DO 90I=1,N  
WRITE(6,101)(P(I,J),J=1,N)  
90 CONTINUE  
DO 55 I=1,N  
DO 55 J=1,N  
55 A(I,J)=0  
DO 56 M=1,N  
DO 57 I=1,N  
J=0  
DO 58 K=NX,NY  
J=J+1  
58 A(M,I)=S(M,K)*P(J,I)+A(M,I)  
57 CONTINUE  
56 CONTINUE  
WRITE(6,222)  
DO 59 I=1,N  
59 WRITE(6,101)(A(I,J),J=1,N)  
101 FORMAT(3X1P4E20.6)  
LX=LX-1  
IF(LX)999,999,998  
998 GO TO 997  
999 STOP  
END
```

MATRIX A

-5.50988	1.87009	0.42291	0.00881
0.28786	-11.81170	5.71190	0.05872
0.04910	4.30803	-12.07070	0.22933
0.00623	0.26985	1.39737	-17.59621

INVERSE MATRIX

1.000000E+00	0.0	0.0	0.0
0.0	1.000000E+00	0.0	0.0
6.234996E-03	2.698510E-01	1.397370E+00	-1.759619E-02
0.0	0.0	0.0	1.000000E+00

PRODUCT MATRIX

-5.511766E+00	1.788420E+00	3.026456E-01	5.334228E+04
2.623788E-01	-1.291474E+01	4.087605E+00	7.198506E-04
1.859188E-01	6.046163E+00	-2.946194E+01	-2.084559E-05
3.725290E-09	5.960464E-08	9.999999E-01	-1.525879E-05

INVERSE MATRIX

1.000000E+00	0.0	0.0	0.0
1.859188E-01	6.046164E+00	-2.946194E+01	-2.082558E+02
0.0	0.0	1.000000E+00	0.0
0.0	0.0	0.0	1.000000E+00

PRODUCT MATRIX

-5.566760E+00	2.957941E-01	9.017315E+00	6.699422E+01
2.952511E+00	-4.232167E+01	-5.625576E+02	-2.244458E+03
5.960464E-08	9.999998E-01	-1.525879E-05	-9.155273E-05
1.892454E-09	9.858258E-09	1.000000E+00	-1.320378E-05

2.4. Program for Evaluation of Characteristic Values
and Characteristic Vectors

INVERSE MATRIX

2.952511E+00	-4.232167E+01	-5.625576E+02	-2.244458E+03
0.0	1.000000E00	0.0	0.0
0.0	0.0	1.000000E+00	0.0
0.0	0.0	0.0	1.000000E+00

PRODUCT MATRIX

-4.788841E+01	-7.972781E+02	-5.349441E+03	-1.229648E+04
9.99999E-01	-1.625879E-05	-2.441406E-04	-2.441406E-04
2.018788E-08	1.000000E+00	-3.901999E-06	-4.624210E-05
6.409642E-10	3.698494E-08	1.000000E+00	-1.176516E-05

square matrix ($n \times n$). This equation may be viewed

as a transformation of the vector $\{x\}$ into the

vector $\{y\}$. The dilation transformation

maps the vector $\{x\}$ into a constant times itself; it

follows that

$$\{y\} = \lambda \{x\} \quad (2.4.2)$$

$$\text{or} \quad [A - \lambda D] \{x\} = \{0\} \quad (2.4.3)$$

where λ is defined as the characteristic value and $\{x\}$ is defined as the associated characteristic vector. The homogeneous Equation (2.4.3) has a solution which exists if and only if, the following determinant

2.4 Program for Evaluation of Characteristic Values and Characteristic Vectors

The characteristic-value, characteristic-vector problem is an extremely important one since the dynamic behavior of a linear mechanical systems are directly predictable by its usage.

Consider the vector matrix equation

$$\{y\} = [A]\{x\} \quad (2.4.1)$$

where $\{y\}$ and $\{x\}$ are column vectors and $[A]$ is a square matrix ($n \times n$). This equation may be viewed as a transformation of the vector $\{x\}$ into the vector $\{y\}$. The dilatation transformation maps the vector $\{x\}$ into a constant times itself; it follows that

$$\{y\} = \lambda [I] \{x\} \quad (2.4.2)$$

or $[[A] - \lambda [I]] \{x\} = \{0\} \quad (2.4.3)$

where λ is defined as the characteristic value and $\{x\}$ is defined as the associated characteristic vector. The homogeneous Equation (2.4.3) has a solution which exists if and only if, the following determinant

equation holds:

$$\det [[A] - \lambda [I]] = 0 \quad (2.4.4)$$

for $n = 3$.

$$\begin{vmatrix} a_{11} - \lambda & a_{12} & a_{13} \\ a_{21} & a_{22} - \lambda & a_{23} \\ a_{31} & a_{32} & a_{33} - \lambda \end{vmatrix} = 0 \quad (2.4.5)$$

This equation of degree 3 (generally of degree n) for λ is called characteristic equation and takes the form

$$\lambda^3 - I_1 \lambda^2 + I_2 \lambda - I_3 = 0 \quad (2.4.6)$$

For each of the roots λ_i , Equation (2.4.3) has a solution $\{x\} \neq 0$ called the characteristic vector of $[A]$. The characteristic values and characteristic vectors solutions are programed by using an iteration process. The characteristic value and the characteristic vectors are related as follows:

$$[[A] - \lambda [I]] \{s\} = \{0\} \quad (2.4.7)$$

This matrix equation can be written as

$$\begin{bmatrix} a_{11} - \lambda & a_{12} & a_{13} & \dots & a_{1n} \\ a_{21} & a_{22} - \lambda & a_{23} & \dots & a_{2n} \\ a_{31} & a_{32} & a_{33} - \lambda & \dots & a_{3n} \\ \vdots & \vdots & \vdots & \ddots & \vdots \\ \vdots & \vdots & \vdots & \ddots & \vdots \\ a_{n1} & a_{n2} & a_{n3} & \dots & a_{nn} - \lambda \end{bmatrix} \begin{Bmatrix} s_1 \\ s_2 \\ s_3 \\ \vdots \\ s_n \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \\ 0 \\ \vdots \\ 0 \end{Bmatrix}$$

(2.4.8)

where λ is the characteristic value, a_{ij} 's are the elements of matrix $[A]$, and s_i 's are the elements of the characteristic vector corresponding to the value λ . The characteristic values and characteristic vectors can be obtained by method of iteration process. Using Equation (2.4.8) form

$$\begin{bmatrix} a_{11} & a_{12} & a_{13} & \dots & a_{1n} \\ a_{21} & a_{22} & a_{23} & \dots & a_{2n} \\ a_{31} & a_{32} & a_{33} & \dots & a_{3n} \\ \vdots & \vdots & \vdots & \ddots & \vdots \\ a_{n1} & a_{n2} & a_{n3} & \dots & a_{nn} \end{bmatrix} \begin{Bmatrix} s_1 \\ s_2 \\ s_3 \\ \vdots \\ s_n \end{Bmatrix} = \lambda \begin{Bmatrix} s_1 \\ s_2 \\ s_3 \\ \vdots \\ s_n \end{Bmatrix}$$

(2.8.9)

In the matrix iteration method, an arbitrary set of s_i (i.e. $s_1, s_2, s_3 \dots s_n$) are used to initiate the problem. For convenience in calculations, s_n is taken equal to unity. Equation (2.4.8) forms a set of homogeneous equations, hence, the absolute values of s 's can not be determined. However, the ratios of s 's may be obtained. The initial set of s_i , $i = 1, 2, 3, \dots n$, are substituted into the left hand side of Equation (2.4.9) performing the indicated matrix multiplication on the left hand side of Equation (2.4.9), the vector on the right hand side is calculated. This vector is factored by defining the new value of s_n as $s_n = \lambda(1)$. This value λ is factored from each of the remaining vector components. The value λ is the first approximation to the characteristic value and the factored vector is the first approximation to the characteristic vector. The method proceeds by taking the next approximation for the vector solution as the previous solution. It is necessary to iterate a number of times in order to improve accuracy.

Continuing this process, the iteration converges, resulting in the characteristic value λ and the

corresponding characteristic vector. The rate of convergence of this iteration process depends on the numerical separation of the characteristic value of matrix [A]. It can also be shown that the characteristic value obtained by this method is the largest characteristic value or equivalently the largest root of the characteristic equation. For the special of a symmetric matrix [A], the characteristic values are always real, and the characteristic vectors are always orthogonal.

The basic flow chart for the program of obtaining characteristic value and characteristic vector by iteration process is shown in Figure 4.



Figure 4. Flow chart of Characteristic value and Characteristic vector

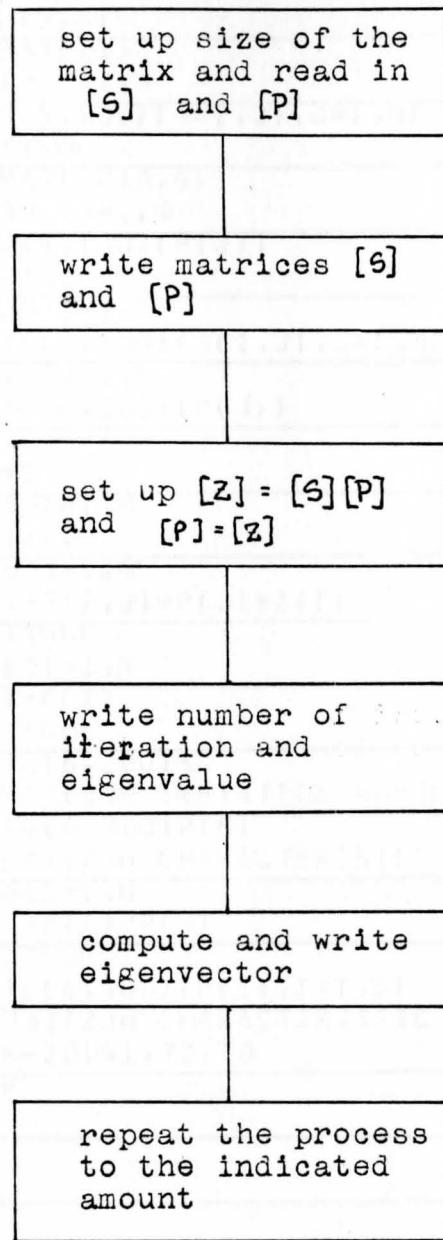


Figure 4 Flow chart of Characteristic value and Characteristic vector

C PROGRAM TO EVALUATE EIGENVALUE & EIGENVECTOR
DIMENSION S(100,100),P(100),Z(100)
READ(5,2)N
2 FORMAT(I1)
DO 10 I=1,N
READ(5,100)(S(I,J),J=1,N)
10 CONTINUE
100 FORMAT(8F10.4)
DO 30 J=1,N
READ(5,100)(P(J))
30 CONTINUE
DO 20 I=1,N
20 WRITE(6,100)(S(I,J),J=1,N)
DO 40 J=1,N
40 WRITE(6,100)(P(J))
K=0
41 K=K+1
DO 50 I=1,N
Z(I)=0.
DO 50 J=1,N
Z(I)=S(I,J)*P(J)+Z(I)
50 CONTINUE
DO 60 I=1,N
P(I)=Z(I)
60 CONTINUE
WRITE(6,200)K
200 FORMAT(20H IRETATION NUMBER= I6)
WRITE(6,300)P(N)
300 FORMAT(25H CHARACTERISTIC VALUE= E10.5)
DO 61 I=1,N
P(I)=P(I)/P(N)
61 CONTINUE
WRITE(6,400)(P(I),I=1,3)
400 FORMAT(23H CHARACTERISTIC VECTOR / (6XE15.5))
IF(K-20)41,70,70
70 STOP
END

2.
-1.
0.

-1.
2.
-1.

0.
-1.
1.

39

1.
-1.
1.

CHARACTERISTIC VECTOR
2.0000 -1.0000 0.0
-1.0000 2.0000 -1.0000
0.0 -1.0000 1.0000

IRETATION NUMBER= 1
CHARACTERISTIC VALUE= 0.200E+01
CHARACTERISTIC VECTOR

0.150E+01
-0.200E+01
0.100E+01

IRETATION NUMBER= 2
CHARACTERISTIC VALUE= 0.300E+01
CHARACTERISTIC VECTOR

0.167E+01
-0.217E+01
0.100E+01

IRETATION NUMBER= 3
CHARACTERISTIC VALUE= 0.317E+01
CHARACTERISTIC VECTOR

0.174E+01
-0.221E+01
0.100E+01

IRETATION NUMBER= 4
CHARACTERISTIC VALUE= 0.321E+01
CHARACTERISTIC VECTOR

0.177E+01
-0.223E+01
0.100E+01

IRETATION NUMBER= 5
CHARACTERISTIC VALUE= 0.323E+01
CHARACTERISTIC VECTOR

0.179E+01
-0.224E+01
0.100E+01

IRETATION NUMBER= 6
CHARACTERISTIC VALUE= 0.324E+01
CHARACTERISTIC VECTOR

0.179E+01
-0.224E+01
0.100E+01

IRETATION NUMBER= 7
CHARACTERISTIC VALUE= 0.324E+01
CHARACTERISTIC VECTOR

0.180E+01
-0.225E+01

0.180E+01
-0.225E+01
0.100E+01
IRETATION NUMBER= 9 CHARACTERISTIC VALUE= 0.325E+01
CHARACTERISTIC VECTOR
0.180E+01
-0.225E+01
0.100E+01
IRETATION NUMBER= 10 CHARACTERISTIC VALUE= 0.325E+01
CHARACTERISTIC VECTOR
0.180E+01
-0.225E+01
0.100E+01
IRETATION NUMBER= 11 CHARACTERISTIC VALUE= 0.325E+01
CHARACTERISTIC VECTOR
0.180E+01
-0.225E+01
0.100E+01
IRETATION NUMBER= 12 CHARACTERISTIC VALUE= 0.325E+01
CHARACTERISTIC VECTOR
0.180E+01
-0.225E+01
0.100E+01
IRETATION NUMBER= 13 CHARACTERISTIC VALUE= 0.325E+01
CHARACTERISTIC VECTOR
0.180E+01
-0.225E+01
0.100E+01
IRETATION NUMBER= 14 CHARACTERISTIC VALUE= 0.325E+01
CHARACTERISTIC VECTOR
0.180E+01
-0.225E+01
0.100E+01
IRETATION NUMBER= 15 CHARACTERISTIC VALUE= 0.325E+01
CHARACTERISTIC VECTOR
0.180E+01
-0.225E+01
0.100E+01
IRETATION NUMBER= 16 CHARACTERISTIC VALUE= 0.325E+01
CHARACTERISTIC VECTOR
0.180E+01
-0.225E+01
0.100E+01
IRETATION NUMBER= 17 CHARACTERISTIC VALUE= 0.325E+01
CHARACTERISTIC VECTOR
0.180E+01
-0.225E+01
0.100E+01
IRETATION NUMBER= 18 CHARACTERISTIC VALUE= 0.325E+01
CHARACTERISTIC VECTOR
0.180E+01

CHARACTERISTIC VECTOR
 0.180E+01
 -0.225E+01
 0.100E+01
 EIGENVALUE NUMBER = 20
 CHARACTERISTIC VALUE = 0.325E+01
 CHARACTERISTIC VECTOR
 0.180E+01
 -0.225E+01
 0.100E+01

2.5 The Method of Cholesky Transformation

R;

The Cholesky process is particularly useful in problems of structural dynamics⁽⁴⁾. This scheme for the solution of a system of linear equations related to structural analysis is very desirable. Theoretically, Cholesky's method is based on the fact that any square matrix may be expressed as the product of an upper-triangular matrix and a lower-triangular matrix.

Let $[A]$ be a non symmetric square matrix, $[U]$ an upper triangular matrix, and $[L]$ a lower-triangular matrix, such that they are related by the following general matrix equation⁽³⁾,

$$[A] = [L][U] \quad (2.5.1)$$

If the square matrix $[A]$ is a (4×4) matrix, then

$$\begin{bmatrix} a_{11} & a_{12} & a_{13} & a_{14} \\ a_{21} & a_{22} & a_{23} & a_{24} \\ a_{31} & a_{32} & a_{33} & a_{34} \\ a_{41} & a_{42} & a_{43} & a_{44} \end{bmatrix} = \begin{bmatrix} 1 & 0 & 0 & 0 \\ l_{21} & 1 & 0 & 0 \\ l_{31} & l_{32} & 1 & 0 \\ l_{41} & l_{42} & l_{43} & 1 \end{bmatrix} \begin{bmatrix} u_{11} & u_{12} & u_{13} & u_{14} \\ 0 & u_{22} & u_{23} & u_{24} \\ 0 & 0 & u_{33} & u_{34} \\ 0 & 0 & 0 & u_{44} \end{bmatrix} \quad (2.5.2)$$

2.5 Program for the Method of Cholesky Transformation

The Cholesky process is particularly useful in problems of structural dynamics⁽⁴⁾. This scheme for the solution of a system of linear equations related to structural analysis is very desirable. Theoretically, Cholesky's method is based on the fact that any square matrix may be expressed as the product of an upper-triangular matrix and a lower-triangular matrix.

Let $[A]$ be a non symmetric square matrix, $[U]$ an upper triangular matrix, and $[L]$ a lower triangular matrix, such that they are related by the following general matrix equation⁽³⁾:

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The expansion of the above equation yields a set of sixteen equations from which the values of the elements of matrices $[L]$ and $[U]$ are obtained as functions of the elements of the matrix $[A]$. To illustrate the procedure, consider the following partial set of equation:

$$a_{11} = u_{11}$$

$$a_{21} = l_{21} u_{11}; \quad l_{21} = \frac{a_{21}}{u_{11}} = \frac{a_{21}}{a_{11}}$$

$$a_{31} = l_{31} u_{11}; \quad l_{31} = \frac{a_{31}}{u_{11}} = \frac{a_{31}}{a_{11}}$$

$$a_{41} = l_{41} u_{11}; \quad l_{41} = \frac{a_{41}}{u_{11}} = \frac{a_{41}}{a_{11}}$$

$$a_{12} = u_{12}$$

$$a_{22} = l_{21} u_{12} + u_{22}; \quad u_{22} = a_{22} - l_{21} u_{12} \\ = a_{22} - \frac{a_{21}}{a_{11}} \cdot a_{12}$$

$$a_{32} = l_{31} u_{12} + l_{32} u_{22}; \quad l_{32} = \frac{(a_{32} - l_{31} u_{12})}{u_{22}}$$

For the $(n \times n)$ symmetric matrix shown, the ten

elements of the matrix can be outlined by the

$$a_{42} = l_{41} u_{12} + l_{42} u_{22}; \quad l_{42} = \frac{(a_{42} - l_{41} u_{12})}{u_{22}}$$

The remaining eight values of the λ 's and the μ 's are obtained from the additional remaining equations.

Cholesky's method offers additional advantages for matrices which are symmetric. A symmetric matrix $[A]$ may be written as the product of two triangular matrices, one of them being the transpose of the other in the form

$$[A] = [U]^T [U] \quad (2.5.3a)$$

which is expanded to

$$\begin{bmatrix} a_{11} & a_{12} & a_{13} & a_{14} \\ a_{12} & a_{22} & a_{23} & a_{24} \\ a_{13} & a_{23} & a_{33} & a_{34} \\ a_{14} & a_{24} & a_{34} & a_{44} \end{bmatrix} = \begin{bmatrix} u_{11} & 0 & 0 & 0 \\ u_{12} & u_{22} & 0 & 0 \\ u_{13} & u_{23} & u_{33} & 0 \\ u_{14} & u_{24} & u_{34} & u_{44} \end{bmatrix} \begin{bmatrix} u_{11} & u_{12} & u_{13} & u_{14} \\ 0 & u_{22} & u_{23} & u_{24} \\ 0 & 0 & u_{33} & u_{34} \\ 0 & 0 & 0 & u_{44} \end{bmatrix} \quad (2.5.3b)$$

For the (4×4) sysmetrix matrix shown, the ten elements of the matrix $[U]$ are obtained by the multiplication of the matrices of Equation (2.5.3b)

and in part become

$$a_{11} = u_{11}^2; \quad u_{11} = (a_{11})^{1/2}$$

$$a_{12} = u_{11} u_{12}; \quad u_{12} = a_{12} \cdot \frac{1}{u_{11}} = \frac{a_{12}}{(a_{11})^{1/2}}$$

$$a_{13} = u_{11} u_{13}; \quad u_{13} = a_{13} \cdot \frac{1}{u_{11}} = \frac{a_{13}}{(a_{11})^{1/2}}$$

$$a_{22} = u_{12}^2 + u_{22}^2; \quad u_{22} = (a_{22} - u_{12})^{1/2} = \left[a_{22} - \frac{a_{12}^2}{a_{11}} \right]^{1/2}$$

(2.5.3c)

These operations are generalized for an $(n \times n)$ matrix by the mathematical expression

$$u_{ij} = \frac{a_{ij} - \sum_{k=1}^{i-1} u_{ki} u_{kj}}{u_{ii}} \quad \text{for } \begin{cases} j = i+1, \dots, n \\ j = 2, \dots, n \end{cases} \quad (2.5.4)$$

$$u_{ii} = \frac{a_{ii}}{u_{ii}} \quad \text{for } \begin{cases} i = 1 \\ j = 2, 3, \dots, n \end{cases} \quad (2.5.6)$$

$$u_{ii} = (a_{ii})^{1/2} \quad \text{for } i = 1 \quad (2.5.7)$$

$$u_{ii} = (a_{ii} - \sum_{k=1}^{i-1} u_{ki}^2)^{1/2} \quad \text{for } i = 2, \dots, n \quad (2.5.8)$$

The flow chart of the computer program for Cholesky triangulation is shown in Figure 5.

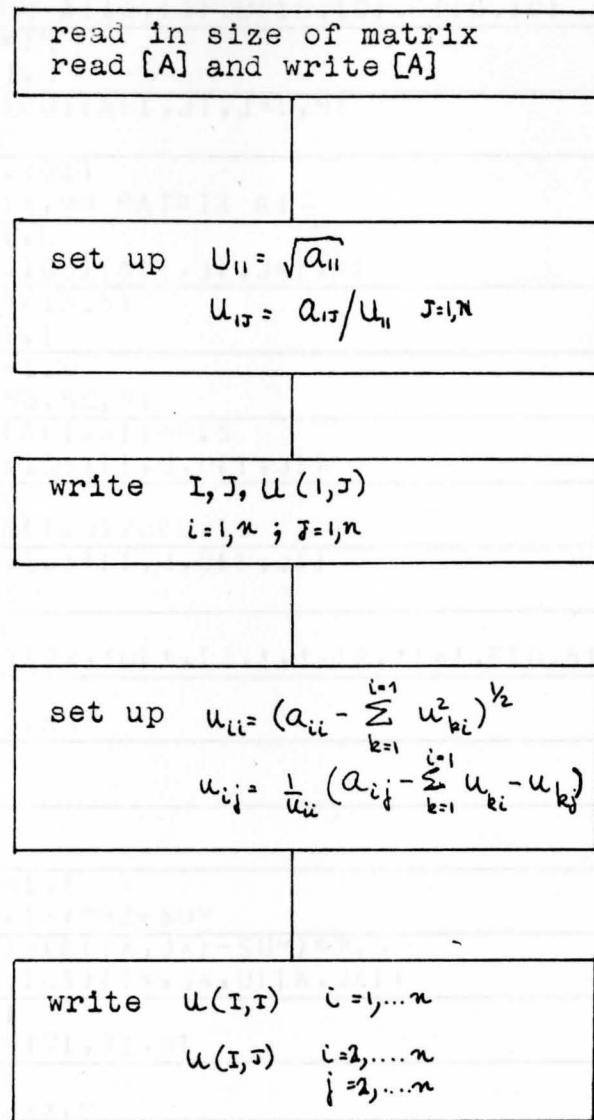


Figure 5 Flow chart of Cholesky Triangulation

```

C      THIS PROGRAM WILL COMPUTE U & U TRANSPOSED-MATRIX FROM
C      THE GIVEN MATRIX
DIMENSION A(10,10),U(10,10),C(10,10)
READ(5,* )N
DO 10I=1,N
READ(5,100)(A(I,J),J=1,N)
10 CONTINUE
WRITE(6,101)
101 FORMAT(1X,9H MATRIX A)
DO 20I=1,N
20 WRITE(6,100)(A(I,J),J=1,N)
100 FORMAT(8F10.5)
DO 30I=1,1
DO 40 J=1,N
IF(J-1)50,50,51
50 U(I,J)=(A(I,J))**.5
WRITE(6,103)(I,J,U(I,J))
GO TO 40
51 U(I,J)=A(I,J)/U(I,I)
WRITE(6,103)(I,J,U(I,J))
40 CONTINUE
30 CONTINUE
103 FORMAT(1(2X,'U(' ,I2,',',',I2,')=' ,F10.5))
NN=N-1
DO 60I=1,NN
SUM=0
J=I
IX=I+1
JX=J+1
DO 70 K=1,I
SUM=U(K,IX)**2+SUM
U(IX,JX)=(A(IX,JX)-SUM)**.5
WRITE(6,103)(IX,JX,U(IX,JX))
JXX=JX+1
IF(JXX-N)71,71,91
71 CONTINUE
DO 75M=JXX,N
SUM=0
DO 80L=1,I
SUM=U(L,JX)*U(L,M)+SUM
U(JX,M)=(A(IX,M)-SUM)/U(IX,JX)
75 WRITE(6,103)(JX,M,U(JX,M))
60 CONTINUE
91 CONTINUE
WRITE(6,104)
104 FORMAT(1X,9H MATRIX U)
DO 90I=1,N
90 WRITE(6,100)(U(I,J),J=1,N)
STOP
END

```

OPTIONS USED - PRINT,MAP,LET,CALL,NORES,NOTERM,SIZE=102400,NAME=***G

NAME	TYPE	ADDR	NAME	TYPE	ADDR	NAME	TYPE	AD
MAIN	SD	150810	IHNNECOMH*	SD	151360	IBCOM#	*	LR 151
IHNCOMH2*	SD	1522F0	SEQDASD *	LR	15271E	IHNLDFIG*	SD	152
IHNFRXPR*	SD	153BC8	FRXPR# *	LR	153BC8	IHNFCVTH*	SD	153
FCVLQUTP*	LR	153E8A	FCVZQUTP*	LR	153FE6	FCVIQUTP*	LR	154
INT6SWCH*	LR	154840	IHNNEFIOS*	SD	1548C0	FIOCS# *	LR	154
IHNNEFNTH*	SD	155E70	ARITH# *	LR	155E70	ADJSWTCH*	LR	156
ERRMON *	LR	156720	IHNERR*	LR	156738	IHNUTABL*	SD	156
ALOG *	LR	156FC0	IHNSEX*	SD	157180	EXP *	LR	157
IHNFCONT*	SD	1577D8	FQCONI# *	LR	1577D8	IHNTRCH*	SD	157
IHNFTEN *	SD	157D68	FTEN# *	LR	157D68			

TOTAL LENGTH 76F0
ENTRY ADDRESS 150810

MATRIX A

1.00000	2.00000	3.00000	2.00000	1.00000
2.00000	5.00000	8.00000	7.00000	6.00000
3.00000	8.00000	17.00000	14.00000	15.00000
2.00000	7.00000	14.00000	23.00000	28.00000
1.00000	6.00000	15.00000	28.00000	62.00000

U(1, 1)= 1.00000

U(1, 2)= 2.00000

U(1, 3)= 3.00000

U(1, 4)= 2.00000

U(1, 5)= 1.00000

U(2, 2)= 1.00000

U(2, 3)= 2.00000

U(2, 4)= 3.00000

U(2, 5)= 4.00000

U(3, 3)= 2.00000

U(3, 4)= 1.00000

U(3, 5)= 2.00000

U(4, 4)= 3.00000

U(4, 5)= 4.00000

U(5, 5)= 5.00000

MATRIX U

1.00000	2.00000	3.00000	2.00000	1.00000
0.0	1.00000	2.00000	3.00000	4.00000
0.0	0.0	2.00000	1.00000	2.00000
0.0	0.0	0.0	3.00000	4.00000
0.0	0.0	0.0	0.0	5.00000

OPTIONS USED - PRINT,MAP,LET,CALL,NORES,NOTERM,SIZE=102400;NAME=**GO

NAME	TYPE	ADDR	NAME	TYPE	ADDR	NAME	TYPE	ADDR	
MAIN	SD	150810	IHNCOMH*	SD	151360	IBCOM#	*	LR	151380
IHNCOMH2*	SD	1522F0	SEQDASD *	LR	15271E	IHNLDFI0*	SD	152A08	
IHNFRXPR*	SD	1538C8	FRXPR# *	LR	153BC8	IHNFCVTH*	SD	153D50	
FCVLQUTP*	LR	153E8A	FCVZOUTP*	LR	153FE6	FCVIOUTP*	LR	15438E	
INT&SWCH*	LR	154840	IHNFIOS*	SD	1548C0	FI0CS# *	LR	1548C0	
IHNFNTH*	SD	155E70	ARITH# *	LR	155E70	ADJSWTC*	LR	156200	
ERRMON *	LR	156720	IHNERR*	LR	156738	IHNUTABL*	SD	156D20	
ALOG *	LR	156FC0	IHNSEXP *	SD	157180	EXP *	LR	157180	
IHNFCONI*	SD	1577D8	FQCONI# *	LR	1577D8	IHNTRCH*	SD	157ACC	
IHNFTEN *	SD	157D68	FTEN# *	LR	157D68				

TOTAL LENGTH 76F0
 ENTRY ADDRESS 150810

MATRIX A

1.00000	2.00000	3.00000	2.00000	3.00000
2.00000	8.00000	8.00000	8.00000	8.00000
3.00000	8.00000	14.00000	12.00000	12.00000
2.00000	8.00000	12.00000	21.00000	16.00000
3.00000	8.00000	12.00000	16.00000	16.00000
U(1, 1)=	1.00000			
U(1, 2)=	2.00000			
U(1, 3)=	3.00000			
U(1, 4)=	2.00000			
U(1, 5)=	3.00000			
U(2, 2)=	2.00000			
U(2, 3)=	1.00000			
U(2, 4)=	2.00000			
U(2, 5)=	1.00000			
U(3, 3)=	2.00000			
U(3, 4)=	2.00000			
U(3, 5)=	1.00000			
U(4, 4)=	3.00000			
U(4, 5)=	2.00000			
U(5, 5)=	1.00000			

MATRIX U

1.00000	2.00000	3.00000	2.00000	3.00000
0.0	2.00000	1.00000	2.00000	1.00000
0.0	0.0	2.00000	2.00000	1.00000
0.0	0.0	0.0	3.00000	2.00000
0.0	0.0	0.0	0.0	1.00000

CHAPTER III

FINITE DIFFERENCE ANALYSIS OF THE RESPONSE
OF PORTAL FRAME3.1 Program for Linear Equations of Motion
of Multi-degree System

The basic equation governing the response on a multi-degree of freedom structure is

$$[M]\{\ddot{x}\} + [K]\{x\} = \{f(t)\} \quad (3.1)$$

where $[M]$ is the mass matrix, $[K]$ is the stiffness matrix, $\{\ddot{x}\}$ is the acceleration vector at any time t , $\{x\}$ is the displacement vector at time t and $\{f(t)\}$ is the disturbing force which varies with time.

A vector iteration technique is utilized to determine the response vector $\{x(t)\}$ for the dynamic system.

The first step is to rewrite Equation (3.1) in the new matrix form

$$\{\ddot{x}\} = [M]^{-1}\{f(t)\} - [M]^{-1}[K]\{x\} \quad (3.2)$$

where $[M]^{-1}$ is assumed to exist.

Recalling the Taylor series expansions of a function in one variable, it follows that

$$f(x) = \frac{f(x_0)}{0!} + \frac{(x-x_0)f'(x_0)}{1!} + \frac{(x-x_0)^2 f''(x_0)}{2!} + \dots \quad (3.3)$$

Using a direct analogy the Tailor series expansion for a time-varying vector becomes

$$\{\ddot{x}_{(t)}\}^{(n+1)} = \{\dot{x}\}^n + \Delta \{\ddot{x}_{(t)}\}^n + \frac{\Delta^2}{2} \{\ddot{\ddot{x}}_{(t)}\}^n + \frac{\Delta^3}{6} \{\ddot{\ddot{\ddot{x}}}_{(t)}\}^n + \dots \quad (3.4)$$

Differentiating Equation (3.4) with respect to time gives

$$\{\dot{x}\}^{(n+1)} = \{\dot{x}\}^n + \Delta \{\ddot{x}\}^n + \frac{\Delta^2}{2} \{\ddot{\ddot{x}}\}^n + \dots \quad (3.5a)$$

$$\{\ddot{x}\}^{(n+1)} = \{\ddot{x}\}^n + \Delta \{\ddot{\ddot{x}}\}^n + \dots \quad (3.5b)$$

where $\Delta = t^{(n+1)} - t^n$

The number of terms in this expansion may at first be arbitrarily chosen. The fewer the number of terms taken, the less accurate the result. The simplest solution may be found by considering no terms on the right hand sides of the expansions which contain derivatives higher than the second. Writing these in reverse order, gives

$$\{\ddot{x}\}^{n+1} = \{\ddot{\ddot{x}}\}^n \quad (3.6a)$$

$$\{\dot{x}\}^{n+1} = \{\dot{\ddot{x}}\}^n \quad (3.6b)$$

$$\{x\}^n = \{x\}^n + \Delta \{\dot{x}\}^n + \frac{\Delta^2}{2} \{\ddot{x}\}^n \quad (3.6c)$$

For a given value of $\{x\}^n$, $\{\dot{x}\}^n$ and $\{\ddot{x}\}^{n+1}$,

$\{\ddot{x}\}^n$ is found directly by use of Equation (3.6a);

$\{\dot{x}\}^{n+1}$ is obtained from Equation (3.6b); $\{x\}^{n+1}$

is calculated from Equation (3.6c). Noting in

Equation (3.6a) that the acceleration at the end of the interval is exactly the same as the acceleration at the beginning of the interval, one defines this procedure as the "constant acceleration method" of iteration (i.e. no derivatives beyond the second is retained). Permutting the value of n to $(n - 1)$ in Equation (3.6a), (3.6b) and (3.6c) yields

$$\{x\}^n = \{x\}^{(n-1)} + \Delta \{\dot{x}\}^{(n-1)} + \frac{\Delta^2}{2} \{\ddot{x}\}^{(n-1)} \quad (3.7a)$$

$$\{\dot{x}\}^n = \{\dot{x}\}^{(n-1)} + \Delta \{\ddot{x}\}^{(n-1)} \quad (3.7b)$$

$$\{\ddot{x}\}^{(n-1)} = \frac{1}{\Delta} \{\dot{x}\}^n - \frac{1}{\Delta} \{\dot{x}\}^{(n-1)} \quad (3.7c)$$

The convenient form of equation (3.12) for calculation purposes is written as

By substituting $\{\ddot{x}\}^{(n-1)}$ in Equation (3.7c) into (3.7a) gives

$$\{x\}^n = \{x\}^{(n-1)} + \Delta \{\dot{x}\}^{(n-1)} + \frac{\Delta}{2} \{\dot{x}\}^n - \frac{\Delta}{2} \{\dot{x}\}^{(n-1)}$$

or in the above equations, the following steps are

$$\{x\}^n = \{x\}^{(n-1)} + \frac{\Delta}{2} \{x\}^{(n-1)} + \frac{\Delta}{2} \{\ddot{x}\}^n$$

or

$$\{x\}^n - \{x\}^{(n-1)} - \frac{\Delta}{2} [\{x\}^{(n-1)} + \{\dot{x}\}^n] = 0 \quad (3.8)$$

Permutting the value of n to $(n + 1)$ Equation (3.8), one obtains

$$\{x\}^{(n+1)} - \{x\}^n - \frac{\Delta}{2} [\{\dot{x}\}^n + \{\ddot{x}\}^{(n+1)}] = 0 \quad (3.9)$$

Subtracting Equation (3.8) from Equation (3.9) gives

$$\{x\}^{(n+1)} - 2\{x\}^n + \{x\}^{(n-1)} = \frac{\Delta}{2} [\{\dot{x}\}^{(n+1)} - \{\dot{x}\}^{(n-1)}] \quad (3.10)$$

From Equation (3.7b) one obtains

$$\{\ddot{x}\}^{(n-1)} = \frac{1}{\Delta} [\{\dot{x}\}^n - \{\dot{x}\}^{(n-1)}] \quad (3.11)$$

Combining Equations (3.10), (3.11) and the permuted form of Equation (3.6a) yields

$$\{\ddot{x}\}^n = \frac{1}{\Delta^2} [\{x\}^{(n+1)} - 2\{x\}^n + \{x\}^{(n-1)}] \quad (3.12)$$

The convenient form of Equation (3.12) for calculation purposes is written as

$$\{x\}^{(n+1)} = 2\{x\}^n - \{x\}^{n-1} + \Delta t^2 \{\ddot{x}\}^n \quad (3.13)$$

Finally at $t = t_n$, it follows from Equation (3.2) that

$$\{\ddot{x}\}^n = -[M]^{-1}[K]\{x\}^n + [M]^{-1}\{f(t)\}^n \quad (3.14)$$

From the above equations, the following steps are taken in the iteration procedure:

$$(1) \text{ At } t=0, n=0, \{x\}^{(0)}=0, \{\ddot{x}\}^{(0)} = [M]^{-1}\{f\}^{(0)}$$

$$\text{and by choice } \{x\}^{(1)} = \frac{\Delta t^2}{2} [M]^{-1}\{f\}^{(0)}$$

$$(2) \text{ At } t=t_1 = \Delta t, n=1$$

$$\{\ddot{x}\}^{(1)} = -[M]^{-1}[K]\{x\}^{(1)} + [M]^{-1}\{f\}^{(1)}$$

and

$$\{x\}^{(2)} = 2\{x\}^{(1)} - \{x\}^{(0)} + \Delta t^2\{\ddot{x}\}^{(1)}$$

$$(3) \text{ At } t=t_2 = 2\Delta t, n=2$$

$$\{\ddot{x}\}^{(2)} = -[M]^{-1}[K]\{x\}^{(2)} + [M]^{-1}\{f\}^{(2)}$$

$$\{x\}^{(3)} = 2\{x\}^{(2)} - \{x\}^{(1)} + \Delta t^2\{\ddot{x}\}^{(2)}$$

$$(4) \text{ At } t=t_i = i(\Delta t), n=i$$

$$\{\ddot{x}\}^{(i)} = -[M]^{-1}[K]\{x\}^{(i)} + [M]^{-1}\{f\}^{(i)}$$

$$\{x\}^{(i+1)} = 2\{x\}^{(i)} - \{x\}^{(i-1)} + \Delta t^2\{\ddot{x}\}^{(i)}$$

Additional steps similar to the latter steps is taken up to the value of n.

The Equation (3.1) may be applied to the solution for a single-bay multi-story frame shown in Figure 3.1

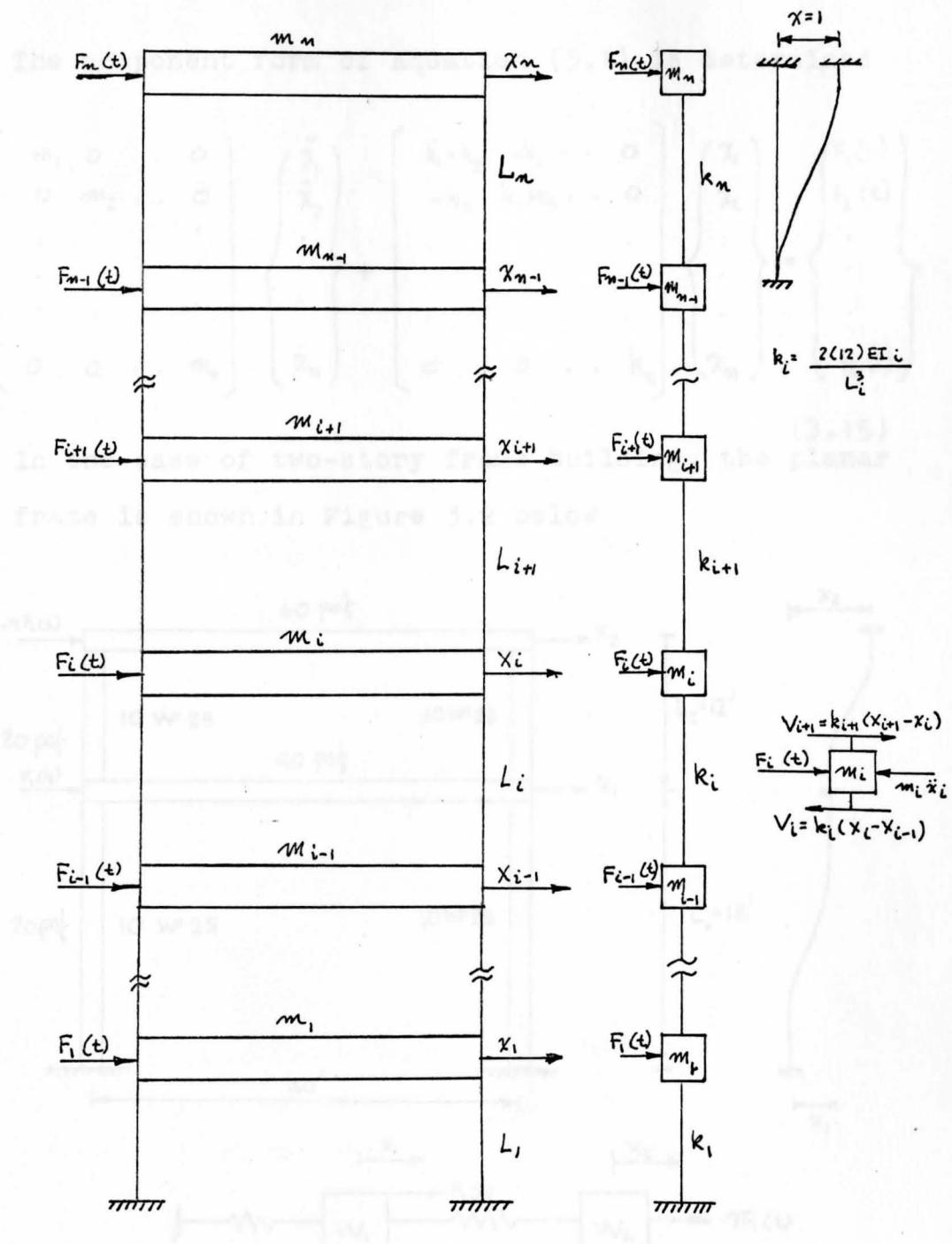


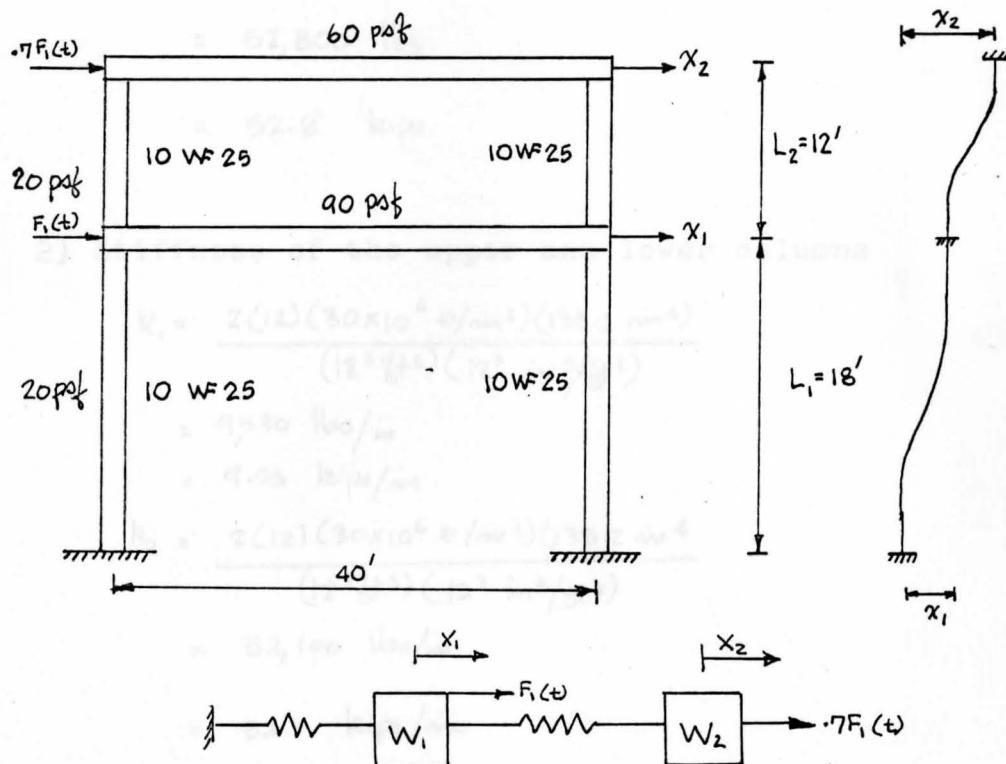
Figure 3.1 Multi-Story Portal Frame

The component form of Equation (3.1) is determined

$$\begin{bmatrix} m_1 & 0 & \dots & 0 \\ 0 & m_2 & \dots & 0 \\ \vdots & \vdots & \ddots & \vdots \\ \vdots & \vdots & \ddots & \vdots \\ 0 & 0 & \dots & m_n \end{bmatrix} \begin{Bmatrix} \ddot{x}_1 \\ \ddot{x}_2 \\ \vdots \\ \vdots \\ \ddot{x}_n \end{Bmatrix} + \begin{bmatrix} k_1+k_2 & -k_2 & \dots & 0 \\ -k_2 & k_2+k_3 & \dots & 0 \\ \vdots & \vdots & \ddots & \vdots \\ \vdots & \vdots & \ddots & \vdots \\ 0 & 0 & \dots & k_n \end{bmatrix} \begin{Bmatrix} x_1 \\ x_2 \\ \vdots \\ \vdots \\ x_n \end{Bmatrix} = \begin{Bmatrix} F_1(t) \\ F_2(t) \\ \vdots \\ \vdots \\ F_n(t) \end{Bmatrix}$$

(3.15)

In the case of two-story frame building, the planar frame is shown in Figure 3.2 below



3) Mass of the upper and lower floors

Figure 3.2 Two-Story Portal Frame

Equation (3.15) reduce to the form

$$\begin{bmatrix} m_1 & 0 \\ 0 & m_2 \end{bmatrix} \begin{Bmatrix} \ddot{x}_1 \\ \ddot{x}_2 \end{Bmatrix} + \begin{bmatrix} k_1+k_2 & -k_2 \\ -k_2 & k_2 \end{bmatrix} \begin{Bmatrix} x_1 \\ x_2 \end{Bmatrix} = \begin{Bmatrix} f_1(t) \\ f_2(t) \end{Bmatrix}$$

The numerical problem is analysed using the computer program with the following numerical input:

1) Weight of the upper and lower floors

$$\begin{aligned} W_1 &= (90 \text{#/ft}^2)(40 \text{ft})(20 \text{ft}) + (2)(20 \text{#/ft}^2)(15 \text{ft})(20 \text{ft}) \\ &= 84,000 \text{ lbs} \\ &= 84.0 \text{ kips.} \end{aligned}$$

$$\begin{aligned} W_2 &= (60 \text{#/ft}^2)(40 \text{ft})(20 \text{ft}) + (2)(20 \text{#/ft}^2)(6 \text{ft})(20 \text{ft}) \\ &= 52,800 \text{ lbs} \\ &= 52.8 \text{ kips.} \end{aligned}$$

2) Stiffness of the upper and lower columns

$$k_1 = \frac{2(12)(30 \times 10^6 \text{#/in}^2)(133.2 \text{in}^4)}{(12^3 \text{ft}^3)(12^3 \text{in}^3/\text{ft}^3)}$$

$$\begin{aligned} &= 9,530 \text{ lbs/in} \\ &= 9.53 \text{ kips/in} \end{aligned}$$

$$k_2 = \frac{2(12)(30 \times 10^6 \text{#/in}^2)(133.2 \text{in}^4)}{(12^3 \text{ft}^3)(12^3 \text{in}^3/\text{ft}^3)}$$

$$\begin{aligned} &= 32,100 \text{ lbs/in} \\ &= 32.1 \text{ kips/in} \end{aligned}$$

3) Mass of the upper and lower floors

$$m_1 = \frac{W_1}{g}$$

$$m_1 = \frac{84.0 \text{ kips}}{(32.2 \text{ ft/sec}^2)(12 \text{ in/ft})}$$

$$= 0.217 \text{ kips-sec}^2/\text{in}$$

$$m_2 = \frac{52.8 \text{ kips}}{(32.2 \text{ ft/sec}^2)(12 \text{ in/ft})}$$

$$= 0.137 \text{ kips-sec}^2/\text{in}$$

4) The natural frequencies and natural periods

$$f_1 = \frac{\omega_1}{2\pi} = 0.812 \text{ Hz}$$

$$f_2 = \frac{\omega_2}{2\pi} = 3.185 \text{ Hz}$$

then

$$\tau_1 = \frac{1}{f_1} = 1.232 \text{ sec.}$$

$$\tau_2 = \frac{1}{f_2} = 0.314 \text{ sec.}$$

5) Dynamic load function on the upper and lower floor

The time variation of the dynamic force $F(t)$
is shown in Figure (3.3)

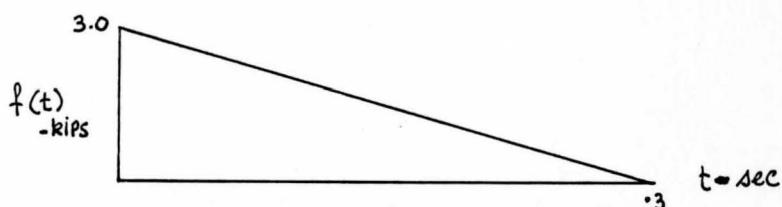


Figure 3.3 Time Variation of Dynamic force

$$f_1 = f(t) = 10t + 3.0$$

$$f_2 = .7f(t) = -7t + 2.7$$

Equation (3.16) becomes

$$\begin{bmatrix} .217 & 0 \\ 0 & .137 \end{bmatrix} \begin{Bmatrix} \ddot{x}_1 \\ \ddot{x}_2 \end{Bmatrix} + \begin{bmatrix} 41.63 & -32.1 \\ -32.1 & 32.1 \end{bmatrix} \begin{Bmatrix} x_1 \\ x_2 \end{Bmatrix} = \begin{Bmatrix} f_1(t) \\ .7f_1(t) \end{Bmatrix}$$

The flow chart of the computer program for the linear equations of motion of the multi-degree of freedom system is shown in Figure (3.4) and the graph of the dynamic responses x_1 and x_2 are shown in Figure (3.5) and figure (3.6)



Figure 3.4 Flow chart of linear equations of motion.

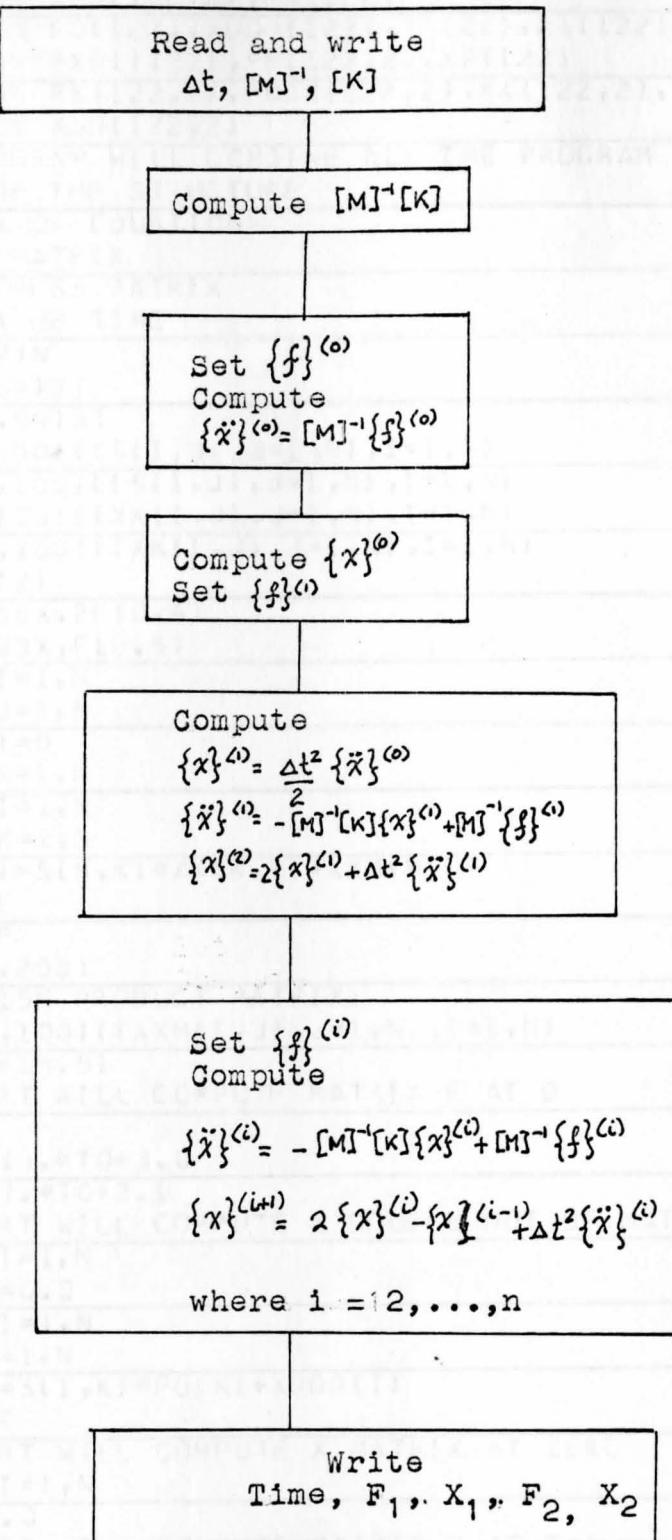


Figure 3.4 Flow chart of linear equations of motion.

DIMENSION XK(122,2),S(72,72)
DIMENSION XXM(122,2),FM(122) 61
DIMENSION FC(122),XDDO(122),T(122),X1(122),F(122),PXDD1(122)
DIMENSION PXD1(122),FF(122,2),XP(122)
DIMENSION PX(122,2),PDDX(122,2),XX(122,2),PXDD(122,2),PXD(122,2)
DIMENSION XDD(122,2)

C THIS PROGRAM WILL COMBINE ALL THE PROGRAM AND GIVE THE FINAL
C RESULT OF THE STRUCTURE
C N=NUMBER OF EQUATIONS
C XM=MASS MATRIX
C XK=STIFFNESS MATRIX
C DT=DELTA OF TIME

READ(5,2)N

READ(5,99)DT

WRITE(6,99)DT

READ(5,100)((S(I,J),J=1,N),I=1,N)

WRITE(6,100)((S(I,J),J=1,N),I=1,N)

READ(5,100)((XK(I,J),J=1,N),I=1,N)

WRITE(6,100)((XK(I,J),J=1,N),I=1,N)

2 FORMAT(I2)

100 FORMAT(30X,2F10.4)

99 FORMAT(30X,F10.5)

DO 110 I=1,N

DO 110 J=1,N

110 XXM(I,J)=0

DO 120 M=1,N

DO 130 I=1,N

DO 140 K=1,N

140 XXM(M,I)=S(M,K)*XK(K,I)+XXM(M,I)

130 CONTINUE

120 CONTINUE

WRITE(6,200)

200 FORMAT(15H PRODUCT MATRIX)

WRITE(6,100)((XXM(I,J),J=1,N),I=1,N)

300 FORMAT(F15.5)

C THIS PART WILL COMPUTE MATRIX F AT 0

T0=0.0

FO(1)=-10.*T0+3.0

FO(2)=-7.*T0+2.1

C THIS PART WILL COMPUTE X DOUBLE DOT OF MATRIX X AT ZERO

DO 180 I=1,N

180 XDDC(I)=0.0

DO 190 I=1,N

DO 220 K=1,N

220 XDDC(I)=S(I,K)*FO(K)+XDDO(I)

190 CONTINUE

C THIS PART WILL COMPUTE X MATRIX AT ZERO

DO 240 I=1,N

240 XP(I)=0.0

C THIS PART WILL COMPUTE MATRIX X AT T=1

DO 260 I=1,N

260 X1(I)=((DT**2.)/2.)*XDDO(I)

C THIS PART WILL COMPUTE X DOUBLE DOT AT T=1

F(1)=-10.*DT+3.0

F(2)=-7.0*DT+2.1

DO 270 M=1,N

PXDD1(M)=0.0

270 PXD1(M)=0.0

DO 280 I=1,N
DO 290 K=1,N

62

PXDD1(I)=XXM(I,K)*X1(K)+PXDD1(I)
PXD1(I)=S(I,K)*F(K)+PXD1(I)

290 CONTINUE

280 CONTINUE

DO 301 I=1,N
L=1

301 XDD(L,I)=-PXDD1(I)+PXD1(I)
C THIS PART WILL COMPUTE MATRIX X AT T=2

M=2

T(M)=2.*DT

DO 310 I=1,N

PX(L,I)=2.*X1(I)

PDDX(L,I)=(DT**2.)*XDD(L,I)

XX(M,I)=PX(L,I)+PDDX(L,I)

310 CONTINUE

L=2

360 CONTINUE

XX(1,1)=X1(1)

XX(1,2)=X1(2)

FF(M,1)=-10.*T(M)+3.0

FF(M,2)=-7.*T(M)+2.10

DO 315 I=1,N

PXDD(L,I)=0.0

315 PXD(L,I)=0.0

DO 320 I=1,N

DO 330 K=1,N

PXDD(L,I)=-XXM(I,K)*XX(M,K)+PXDD(L,I)

PXD(L,I)=S(I,K)*FF(M,I)+PXD(L,I)

330 CONTINUE

320 CONTINUE

DO 340 I=1,N

XDD(L,I)=PXDD(L,I)+PXD(L,I)

340 CONTINUE

C THIS PART WILL COMPUTE MATRIX X FOR ANY TIME

LX=(.3/DT)+1

LY=2*LX

DO 350 I=1,N

J=L+1

JJ=L-1

XX(J,I)=2.*XX(L,I)-XX(JJ,I)+DT**2.*XDD(L,I)

350 WRITE(6,100)XX(J,I)

M=M+1

T(M)=M*DT

L=L+1

400 FORMAT(I3)

IF(L-LX)360,360,370

370 CONTINUE

L=LX

801 CONTINUE

DO 803 I=1,N

803 PXDD(L,I)=0.0

DO 800 I=1,N

DO 810 K=1,N

PXDD(L,I)=-XXM(I,K)*XX(L,K)+PXDD(L,I)

810 CONTINUE

800 CONTINUE

DO 820 I=1,N
FF(L,I)=0.0
J=L+1
JJ=L-1
XX(J,I)=2.*XX(L,I)-XX(JJ,I)+DT**2.*PXDD(L,I)

820 CONTINUE

L=L+1
IF(L-LY)801,801,802

802 CONTINUE

MX0=0

Y=0.0

WRITE(6,500)

500 FORMAT('1',7X,7H NUMBER,9X,5H TIME,12X,3H F1,8X,3H X1,8X,3H F2,5X,
13H X2)

WRITE(6,600)MX0,Y,F0(1),XP(1),F0(2),XP(2)
MX0=1

Y=DT

WRITE(6,600)MX0,Y,F(1),X1(1),F(2),X1(2)

600 FORMAT(9X,I3,10X,F10.5,5X,4F10.5)

DO 700 I=2,LY

Y=I*DT

I1=1

I2=2

700 WRITE(6,600)I,Y,FF(I,I1),XX(I,I1),FF(I,I2),XX(I,I2)

STOP

END

VS LOADER

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OPTIONS USED - PRINT,MAP,LET,CALL,NORES,NOTERM,SIZE=102400,NAME=**GO

NAME	TYPE	ADDR	NAME	TYPE	ADDR	NAME	TYPE	ADDR
MAIN	SD	120810	IHNECOMH*	SD	129DD8	IBCOM# *	LR	129E04
IHNCCMH2*	SD	12AD68	SEQDASD *	LR	12B196	IHNFRXPR*	SD	12B480
ADCON# *	LR	12B608	FCVAOUTP*	LR	12B6B2	FCVLOUTP*	LR	12B742
FCVEOUTP*	LR	12BD38	FCVCOUTP*	LR	12BE82	INT6SWCH*	LR	12C0F8
FIOLCSBEP*	LR	12C17E	IHNFIOS2*	SD	12D170	IHNENFTNTH*	SD	12D728
IHNNUOPT *	SD	12DC70	IHNERRM *	SD	12DF08	ERRMON *	LR	12DFD8
ALOG10 *	LR	12E5D8	ALOG *	LR	12E5F0	IHNSEXP *	SD	12E7B0
FQCONO# *	LR	12E960	IHNFCONI*	SD	12EE08	FQCONI# *	LR	12EF08
IHNTRCH *	LR	12F378	ERRTRA *	LR	12F380	IHNFTEN *	SD	12F620

TOTAL LENGTH EFA8
 ENTRY ADDRESS 120810

0.02000
 4.6082 0.0
 0.0 7.2990

41.6300 -32.1000
 -32.1000 32.1000

PRODUCT MATRIX

191.8394 -147.9232
 -234.2979 234.2979

0.0232

0.0258

0.0400

0.0444

0.0603

0.0671

0.0837

0.0933

0.1095

0.1222

0.1371

0.1532

0.1658

0.1856

0.1950

0.2186

0.2240

0.2513

0.2521

0.2832

0.2788

0.3134

0.3033

0.3411

0.3252

0.3657

0.3437

0.3865

0.3583

0.4038

NUMBER	TIME	F1	X1	F2	X2
0	0.0	3.00000	0.0	2.10000	0.0 65
1	0.00500	2.95000	0.00017	2.06500	0.00019
2	0.01000	2.90000	0.00069	2.03000	0.00076
3	0.01500	2.85000	0.00153	1.99500	0.00170
4	0.02000	2.80000	0.00270	1.96000	0.00300
5	0.02500	2.75000	0.00420	1.92500	0.00466
6	0.03000	2.70000	0.00601	1.89000	0.00666
7	0.03500	2.65000	0.00812	1.85500	0.00901
8	0.04000	2.60000	0.01054	1.82000	0.01169
9	0.04500	2.55000	0.01324	1.78500	0.01470
10	0.05000	2.50000	0.01623	1.75000	0.01802
11	0.05500	2.45000	0.01950	1.71500	0.02165
12	0.06000	2.40000	0.02304	1.68000	0.02558
13	0.06500	2.35000	0.02684	1.64500	0.02981
14	0.07000	2.30000	0.03089	1.61000	0.03431
15	0.07500	2.25000	0.03518	1.57500	0.03909
16	0.08000	2.20000	0.03971	1.54000	0.04414
17	0.08500	2.15000	0.04446	1.50500	0.04944
18	0.09000	2.10000	0.04944	1.47000	0.05498
19	0.09500	2.05000	0.05462	1.43500	0.06076
20	0.10000	2.00000	0.05999	1.40000	0.06677
21	0.10500	1.95000	0.06556	1.36500	0.07299
22	0.11000	1.90000	0.07131	1.33000	0.07942
23	0.11500	1.85000	0.07723	1.29500	0.08604
24	0.12000	1.80000	0.08331	1.26000	0.09285
25	0.12500	1.75000	0.08954	1.22500	0.09983
26	0.13000	1.70000	0.09591	1.19000	0.10698
27	0.13500	1.65000	0.10242	1.15500	0.11427
28	0.14000	1.60000	0.10904	1.12000	0.12171
29	0.14500	1.55000	0.11578	1.08500	0.12928
30	0.15000	1.50000	0.12262	1.05000	0.13697
31	0.15500	1.45000	0.12955	1.01500	0.14476
32	0.16000	1.40000	0.13656	0.98000	0.15265
33	0.16500	1.35000	0.14364	0.94500	0.16063
34	0.17000	1.30000	0.15078	0.91000	0.16868
35	0.17500	1.25000	0.15797	0.87500	0.17679
36	0.18000	1.20000	0.16520	0.84000	0.18495
37	0.18500	1.15000	0.17247	0.80500	0.19314
38	0.19000	1.10000	0.17975	0.77000	0.20137
39	0.19500	1.05000	0.18704	0.73500	0.20960
40	0.20000	1.00000	0.19433	0.70000	0.21784
41	0.20500	0.95000	0.20161	0.66500	0.22607
42	0.21000	0.90000	0.20887	0.63000	0.23428
43	0.21500	0.85000	0.21609	0.59500	0.24245
44	0.22000	0.80000	0.22327	0.56000	0.25058
45	0.22500	0.75000	0.23041	0.52500	0.25864
46	0.23000	0.70000	0.23748	0.49000	0.26664
47	0.23500	0.65000	0.24447	0.45500	0.27456
48	0.24000	0.60000	0.25139	0.42000	0.28239
49	0.24500	0.55000	0.25821	0.38500	0.29011
50	0.25000	0.50000	0.26493	0.35000	0.29771
51	0.25500	0.45000	0.27154	0.31500	0.30518
52	0.26000	0.40000	0.27803	0.28000	0.31252
53	0.26500	0.35000	0.28438	0.24500	0.31970
54	0.27000	0.30000	0.29060	0.21000	0.32672
55	0.27500	0.25000	0.29666	0.17500	0.33257
56	0.28000	0.20000	0.30256	0.14000	0.34024
57	0.28500	0.15000	0.30853	0.10500	0.34716

wn State University	COMPUTER CENTER				
64	0.32000	0.0	0.33409	0.0	0.37576
65	0.32500	0.0	0.33863	0.0	0.38086
66	0.33000	0.0	0.34295	0.0	0.38571
67	0.33500	0.0	0.34706	0.0	0.39031
68	0.34000	0.0	0.35094	0.0	0.39466
69	0.34500	0.0	0.35460	0.0	0.39875
70	0.35000	0.0	0.35803	0.0	0.40258
71	0.35500	0.0	0.36124	0.0	0.40616
72	0.36000	0.0	0.36421	0.0	0.40946
73	0.36500	0.0	0.36695	0.0	0.41251
74	0.37000	0.0	0.36946	0.0	0.41528
75	0.37500	0.0	0.37173	0.0	0.41779
76	0.38000	0.0	0.37377	0.0	0.42003
77	0.38500	0.0	0.37556	0.0	0.42200
78	0.39000	0.0	0.37712	0.0	0.42369
79	0.39500	0.0	0.37843	0.0	0.42511
80	0.40000	0.0	0.37950	0.0	0.42626
81	0.40500	0.0	0.38032	0.0	0.42714
82	0.41000	0.0	0.38124	0.0	0.42806
83	0.41500	0.0	0.38133	0.0	0.42812
84	0.42000	0.0	0.38117	0.0	0.42789
85	0.42500	0.0	0.38077	0.0	0.42740
86	0.43000	0.0	0.38013	0.0	0.42663
87	0.43500	0.0	0.37923	0.0	0.42559
88	0.44000	0.0	0.37810	0.0	0.42427
89	0.44500	0.0	0.37672	0.0	0.42269
90	0.45000	0.0	0.37509	0.0	0.42084
91	0.45500	0.0	0.37322	0.0	0.41872
92	0.45500	0.0	0.37111	0.0	0.41633
93	0.46000	0.0	0.36877	0.0	0.41368
94	0.46500	0.0	0.36618	0.0	0.41076
95	0.47000	0.0	0.36335	0.0	0.40759
96	0.47500	0.0	0.36029	0.0	0.40415
97	0.48000	0.0	0.35700	0.0	0.40046
98	0.48500	0.0	0.35347	0.0	0.39651
99	0.49000	0.0	0.34972	0.0	0.39231
100	0.49500	0.0	0.34574	0.0	0.38786
101	0.50000	0.0	0.34153	0.0	0.38317
102	0.50500	0.0	0.33711	0.0	0.37823
103	0.51000	0.0	0.33246	0.0	0.37305
104	0.51500	0.0	0.32760	0.0	0.36763
105	0.52000	0.0	0.32253	0.0	0.36198
106	0.52500	0.0	0.31725	0.0	0.35609
107	0.53000	0.0	0.31177	0.0	0.34993
108	0.53500	0.0	0.30609	0.0	0.34365
109	0.54000	0.0	0.30020	0.0	0.33709
110	0.54500	0.0	0.29413	0.0	0.33032
111	0.55000	0.0	0.28787	0.0	0.32334
112	0.55500	0.0	0.28142	0.0	0.31615
113	0.56000	0.0	0.27479	0.0	0.30875
114	0.56500	0.0	0.26798	0.0	0.30116
115	0.57000	0.0	0.26101	0.0	0.29337
116	0.57500	0.0	0.25386	0.0	0.28540
117	0.58000	0.0	0.24656	0.0	0.27723
118	0.58500	0.0	0.23909	0.0	0.26889
119	0.59000	0.0	0.23148	0.0	0.26038
120	0.59500	0.0	0.22372	0.0	0.25169
121	0.60000	0.0	0.21581	0.0	0.24284
122	0.60500	0.0	0.20777	0.0	0.23383
	0.61000	0.0	0.19959	0.0	0.22467

NUMBER	TIME	F1	X1	F2	X2
0	0.0	3.00000	0.0	2.10000	0.0
1	0.01000	2.90000	0.00069	2.03000	0.00077
2	0.02000	2.80000	0.00272	1.96000	0.00301
3	0.03000	2.70000	0.00603	1.89000	0.00668
4	0.04000	2.60000	0.01056	1.82000	0.01172
5	0.05000	2.50000	0.01627	1.75000	0.01805
6	0.06000	2.40000	0.02308	1.68000	0.02562
7	0.07000	2.30000	0.03093	1.61000	0.03436
8	0.08000	2.20000	0.03976	1.54000	0.04419
9	0.09000	2.10000	0.04949	1.47000	0.05505
10	0.10000	2.00000	0.06006	1.40000	0.06684
11	0.11000	1.90000	0.07138	1.33000	0.07950
12	0.12000	1.80000	0.08339	1.26000	0.09294
13	0.13000	1.70000	0.09600	1.19000	0.10707
14	0.14000	1.60000	0.10914	1.12000	0.12182
15	0.15000	1.50000	0.12272	1.05000	0.13708
16	0.16000	1.40000	0.13666	0.98000	0.15277
17	0.17000	1.30000	0.15089	0.91000	0.16881
18	0.18000	1.20000	0.16533	0.84000	0.18508
19	0.19000	1.10000	0.17988	0.77000	0.20151
20	0.20000	1.00000	0.19446	0.70000	0.21799
21	0.21000	0.90000	0.20900	0.63000	0.23443
22	0.22000	0.80000	0.22342	0.56000	0.25074
23	0.23000	0.70000	0.23763	0.49000	0.26681
24	0.24000	0.60000	0.25154	0.42000	0.28256
25	0.25000	0.50000	0.26509	0.35000	0.29789
26	0.26000	0.40000	0.27819	0.28000	0.31270
27	0.27000	0.30000	0.29076	0.21000	0.32691
28	0.28000	0.20000	0.30273	0.14000	0.34043
29	0.29000	0.10000	0.31402	0.07000	0.35316
30	0.30000	0.0	0.32455	0.0	0.36503
31	0.31000	0.0	0.33426	0.0	0.37596
32	0.32000	0.0	0.34312	0.0	0.38590
33	0.33000	0.0	0.35110	0.0	0.39484
34	0.34000	0.0	0.35819	0.0	0.40276
35	0.35000	0.0	0.36436	0.0	0.40963
36	0.36000	0.0	0.36961	0.0	0.41544
37	0.37000	0.0	0.37391	0.0	0.42018
38	0.38000	0.0	0.37725	0.0	0.42383
39	0.39000	0.0	0.37962	0.0	0.42640
40	0.40000	0.0	0.38102	0.0	0.42786
41	0.41000	0.0	0.38144	0.0	0.42823
42	0.42000	0.0	0.38087	0.0	0.42750
43	0.43000	0.0	0.37932	0.0	0.42568
44	0.44000	0.0	0.37679	0.0	0.42278
45	0.45000	0.0	0.37329	0.0	0.41879
46	0.46000	0.0	0.36882	0.0	0.41374
47	0.47000	0.0	0.36340	0.0	0.40764
48	0.48000	0.0	0.35703	0.0	0.40050
49	0.49000	0.0	0.34974	0.0	0.39234
50	0.50000	0.0	0.34154	0.0	0.38319
51	0.51000	0.0	0.33246	0.0	0.37305
52	0.52000	0.0	0.32252	0.0	0.36197
53	0.53000	0.0	0.31175	0.0	0.34997
54	0.54000	0.0	0.30017	0.0	0.33706
55	0.55000	0.0	0.28782	0.0	0.32330
56	0.56000	0.0	0.27474	0.0	0.30870
57	0.57000	0.0	0.26224	0.0	0.29323

NUMBER	TIME	F1	X1	F2	X2
0	0.0	3.00000	0.0	2.10000	0.0
1	0.02000	2.80000	0.00276	1.96000	0.00307
2	0.04000	2.60000	0.01066	1.82000	0.01183
3	0.06000	2.40000	0.02323	1.68000	0.02579
4	0.08000	2.20000	0.03997	1.54000	0.04442
5	0.10000	2.00000	0.06032	1.40000	0.06713
6	0.12000	1.80000	0.08370	1.26000	0.09328
7	0.14000	1.60000	0.10950	1.12000	0.12222
8	0.16000	1.40000	0.13708	0.98000	0.15324
9	0.18000	1.20000	0.16578	0.84000	0.18560
10	0.20000	1.00000	0.19496	0.70000	0.21856
11	0.22000	0.80000	0.22395	0.56000	0.25135
12	0.24000	0.60000	0.25210	0.42000	0.28320
13	0.26000	0.40000	0.27878	0.28000	0.31337
14	0.28000	0.20000	0.30333	0.14000	0.34111
15	0.30000	0.00000	0.32517	0.00000	0.36573
16	0.32000	0.0	0.34369	0.0	0.38654
17	0.34000	0.0	0.35871	0.0	0.40333
18	0.36000	0.0	0.37007	0.0	0.41594
19	0.38000	0.0	0.37764	0.0	0.42426
20	0.40000	0.0	0.38134	0.0	0.42820
21	0.42000	0.0	0.38111	0.0	0.42775
22	0.44000	0.0	0.37694	0.0	0.42293
23	0.46000	0.0	0.36888	0.0	0.41380
24	0.48000	0.0	0.35699	0.0	0.40046
25	0.50000	0.0	0.34141	0.0	0.38305
26	0.52000	0.0	0.32229	0.0	0.36173
27	0.54000	0.0	0.29984	0.0	0.33672
28	0.56000	0.0	0.27431	0.0	0.30825
29	0.58000	0.0	0.24597	0.0	0.27660
30	0.60000	0.0	0.21512	0.0	0.24208
31	0.62000	0.0	0.18208	0.0	0.20503
32	0.64000	0.0	0.14721	0.0	0.16583

NUMBER	TIME	F1	X1	F2	X2
0	0.0	3.00000	0.0	2.10000	0.0
1	0.03000	2.70000	0.00622	1.89000	0.00690
2	0.06000	2.40000	0.02348	1.68000	0.02607
3	0.09000	2.10000	0.05012	1.47000	0.05573
4	0.12000	1.80000	0.08423	1.26000	0.09386
5	0.15000	1.50000	0.12375	1.05000	0.13824
6	0.18000	1.20000	0.16654	0.84000	0.18646
7	0.21000	0.90000	0.21037	0.63000	0.23600
8	0.24000	0.60000	0.25304	0.42000	0.28427
9	0.27000	0.30000	0.29235	0.21000	0.32872
10	0.30000	0.0	0.32619	0.00000	0.36687
11	0.33000	0.0	0.35255	0.0	0.39645
12	0.36000	0.0	0.37083	0.0	0.41677
13	0.39000	0.0	0.38056	0.0	0.42740
14	0.42000	0.0	0.38149	0.0	0.42815
15	0.45000	0.0	0.37355	0.0	0.41907
16	0.48000	0.0	0.35691	0.0	0.40038
17	0.51000	0.0	0.33194	0.0	0.37253
18	0.54000	0.0	0.29927	0.0	0.33612
19	0.57000	0.0	0.25967	0.0	0.29194
20	0.60000	0.0	0.21410	0.0	0.24095
21	0.63000	0.0	0.16365	0.0	0.18430
22	0.66000	0.0	0.10947	0.0	0.12330

NUMBER	TIME	F1	X1	F2	X2	70
0	0.0	3.00000	0.0	2.10000	0.0	
1	0.05000	2.50000	0.01728	1.75000	0.01916	
2	0.10000	2.00000	0.06216	1.40000	0.06915	
3	0.15000	1.50000	0.12584	1.05000	0.14060	
4	0.20000	1.00000	0.19844	0.70000	0.22256	
5	0.25000	0.50000	0.26970	0.35000	0.30317	
6	0.30000	0.00000	0.32948	0.00000	0.37056	
7	0.35000	0.0	0.36828	0.0	0.41389	
8	0.40000	0.0	0.38351	0.0	0.43050	
9	0.45000	0.0	0.37401	0.0	0.41959	
10	0.50000	0.0	0.34031	0.0	0.38198	
11	0.55000	0.0	0.28465	0.0	0.31997	
12	0.60000	0.0	0.21080	0.0	0.23727	
13	0.65000	0.0	0.12360	0.0	0.13906	
14	0.70000	0.0	0.02854	0.0	0.03180	

Figure 3.5 Oscillating graph of displacement of X.

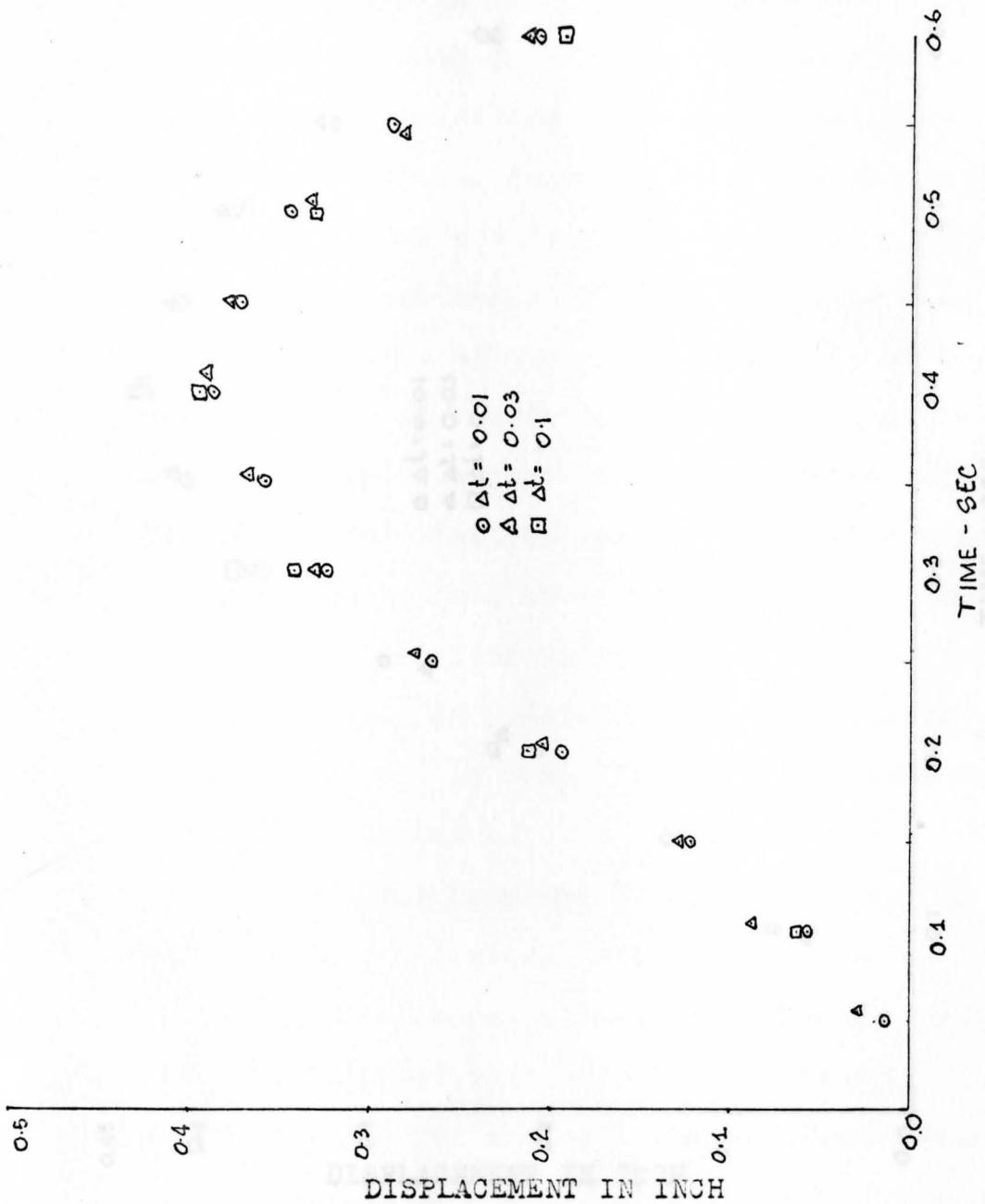


Figure 3.5 Comparable graph of displacement of X_1

CHAPTER IV

DISCUSSION AND COMPUTATION

4.1 Discussion

The programs in Chapter III are intentionally written in FORTRAN language for the dual purpose of illustrating the mathematical operations basic to the matrix formulation of the problem.

This approach allows the reader to follow step by step the mathematical logic involved in the computer solution as well as giving the reader a better understanding of the computation and interaction of the various factors necessary to establish produce the solution.

The hand calculations for any given problem can be performed by hand in a reasonable time for matrices of order three or less. If the order of the matrices is greater than three, the computer solutions offer the most efficient procedure for problem solutions.

The FORTRAN programs developed in Chapter III which include the matrix determinant, matrix inversion, the characteristic equation determination,

the evaluation of characteristic value and the method of velocity

and displacement are comparable to those of the usual

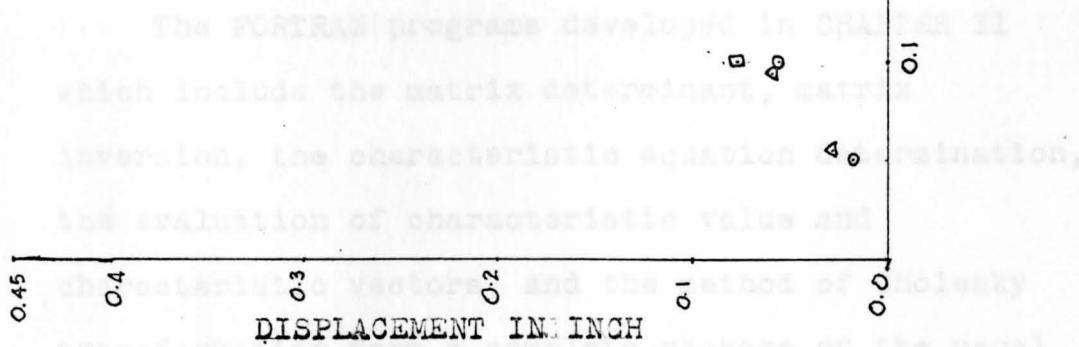


Figure 3.6 Comparable graph of displacement of X_2

CHAPTER IV. TYPES OF STRUCTURES

DISCUSSION AND CONCLUSION

4.1 Discussion

The programs in CHAPTER II are intentionally written in FORTRAN IV language for the distinct purpose of illustrating the mathematical operations basic to the matrix formulation of the problem. The procedure allows the reader to follow step by step the mathematical logic involved in the problem solution as well as giving the reader a basic understanding of the formation and interaction of the necessary equations which produce the solution. The matrix calculations for any given program may be performed by hand in a reasonable time interval for matrices of order three or less. If the order of the matrices is greater than three, the computer solutions offer the most efficient processes for problem solutions.

The FORTRAN programs developed in CHAPTER II which include the matrix determinant, matrix inversion, the characteristic equation determination, the evaluation of characteristic value and characteristic vectors, and the method of Cholesky transformation form a complete package of the usual matrix operations common to the field of Structural

Dynamics. In most cases, analyses of structures subject to dynamic loading usually involve a large number of degree of freedom which are efficiently processed by matrix techniques. Each of the above programs is specifically written to accommodate an arbitrary number of degree of freedom and hence the computer package is useful for the range of problems encountered.

The program in CHAPTER III is also written in FORTRAN IV language. This program illustrates the usefulness of programming techniques to formulate the solution of a family of coupled linear differential equations utilizing finite difference methods. This technique replaces the rather complicated classical functional type solution with a simple numerical iterative method which strictly relies on a vast number of algebraic operations for which the computer is extremely efficient in processing. The degree of accuracy using this method is bases upon only the size limitation of computer.

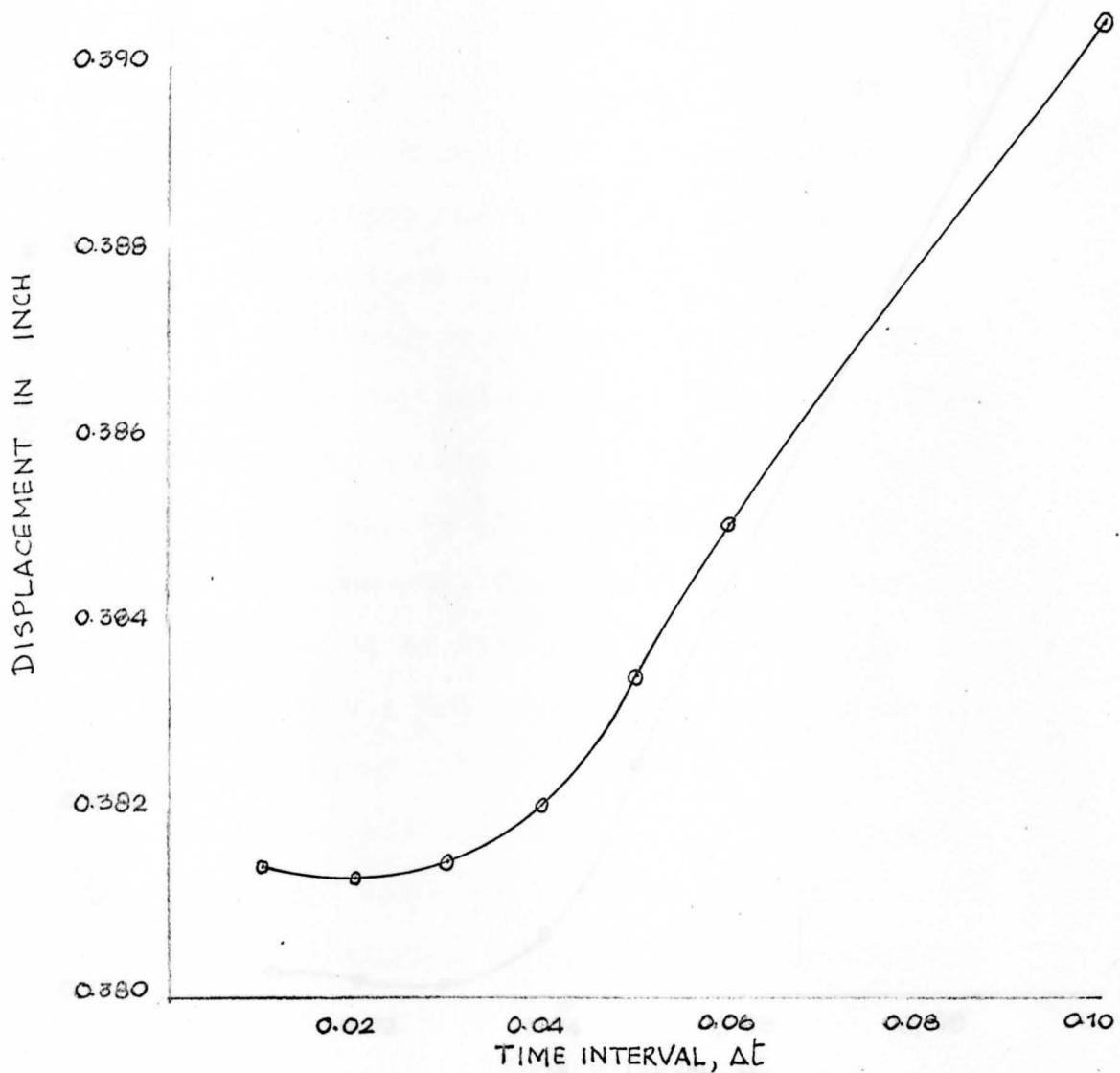
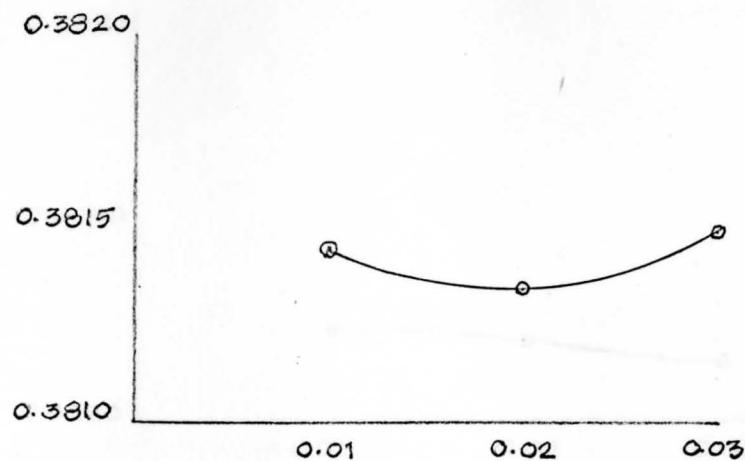


FIGURE 4.1 COMPARABLE GRAPH OF MAXIMUM DISPLACEMENT OF ξ

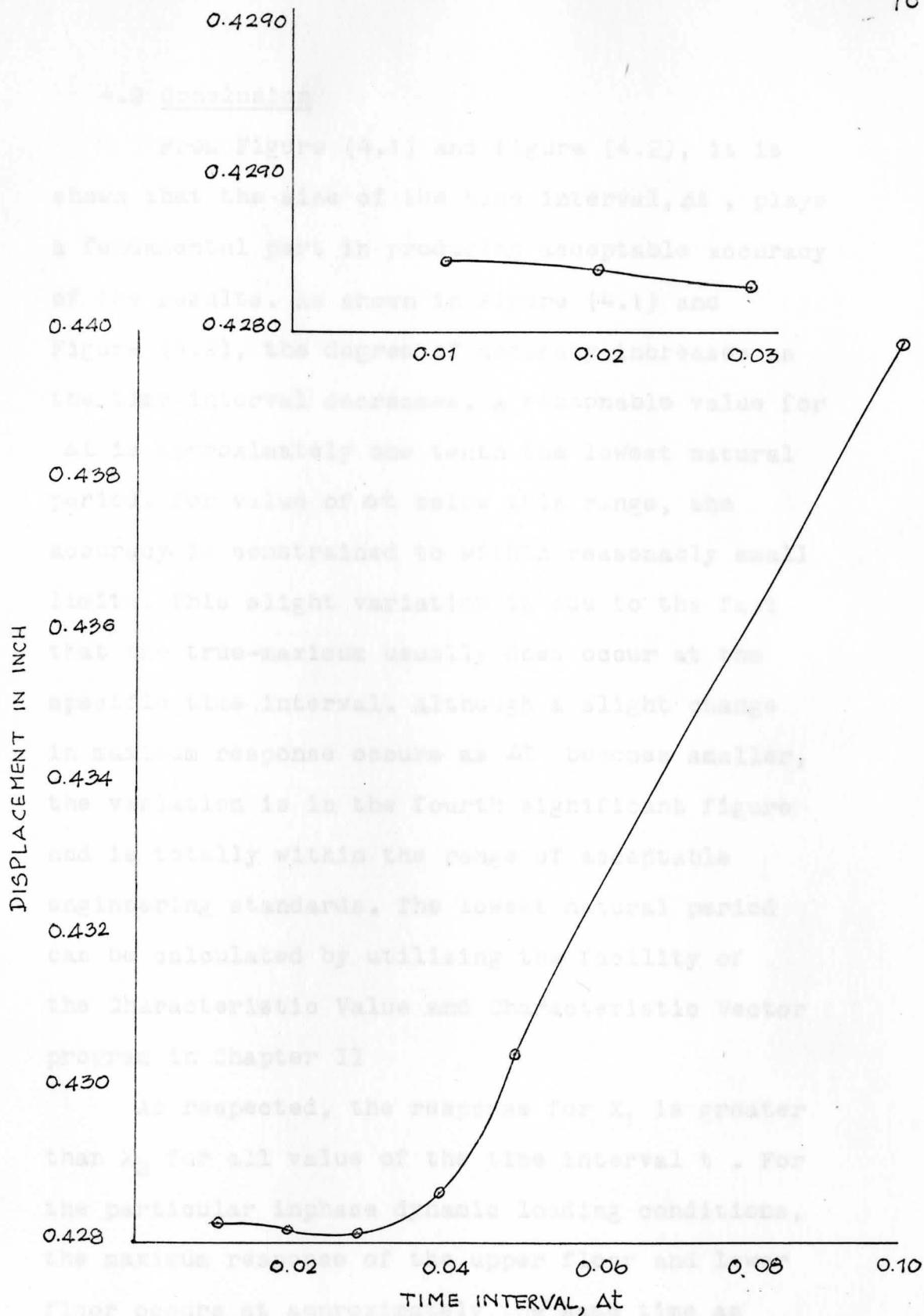


FIGURE 4.2 COMPARABLE GRAPH OF MAXIMUM DISPLACEMENT OF X_2

4.2 Conclusion

From Figure (4.1) and Figure (4.2), it is shown that the size of the time interval, Δt , plays a fundamental part in producing acceptable accuracy of the results. As shown in Figure (4.1) and Figure (4.2), the degree of accuracy increases as the time interval decreases. A reasonable value for Δt is approximately one tenth the lowest natural period. For value of Δt below this range, the accuracy is constrained to within reasonably small limits. This slight variation is due to the fact that the true-maximum usually does occur at the specific time interval. Although a slight change in maximum response occurs as Δt becomes smaller, the variation is in the fourth significant figure and is totally within the range of acceptable engineering standards. The lowest natural period can be calculated by utilizing the facility of the Characteristic Value and Characteristic Vector program in Chapter II.

As respected, the response for X_1 is greater than X_2 for all value of the time interval t . For the particular inphase dynamic loading conditions, the maximum response of the upper floor and lower floor occurs at approximately the same time as

shown in Figure (4.1) and Figure (4.2) which is an expected result.

For sample problem chosen, the maximum upper floor displacement is ≈ 0.38144 inch while the maximum lower floor displacement is ≈ 0.42823 inch. These values are within the acceptable elastic range of the material since the column lengths are 18 feet and 12 feet respectively and column size is a 10 WF 25.

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